



Publication Year	2017
Acceptance in OA	2020-08-31T09:04:13Z
Title	Statistics of impacts among orbiting bodies: a Monte Carlo approach
Authors	DELL'ORO, Aldo
Publisher's version (DOI)	10.1093/mnras/stx402
Handle	http://hdl.handle.net/20.500.12386/26991
Journal	MONTHLY NOTICES OF THE ROYAL ASTRONOMICAL SOCIETY
Volume	467

Statistics of impacts among orbiting bodies: a Monte Carlo approach

Aldo Dell’Oro,^{1*}

¹*INAF, Osservatorio Astrofisico di Arcetri, Largo E. Fermi 5, I-50125, Firenze, Italy*

Accepted XXX. Received YYY; in original form ZZZ

ABSTRACT

In this paper we describe a method to investigate the statistics of collisions among bodies orbiting around a common central mass but not interacting each other, like minor bodies of the Solar System. This method can be used to derive the frequency of collisions and the distribution of any dynamical parameter related to the collision circumstances. It is based on the idea to derive the statistics of impacts from a random sampling of the orbital elements space of bodies under investigation. The fundamental approach is not new, but the mathematical framework is completely original and the final procedure is very simple and easily implementable. The motivation behind this work is to overcome all limitations due to the assumptions about the dynamical behavior of orbits, which the methods developed so far are based on. We show all theoretical details of the method, its practical usage, including the determination of error, showing a set of examples demonstrating a satisfactory agreement with independent approaches, when they are available. Limits and drawbacks of the methodology are also highlighted as well.

Key words: minor planets, asteroids: general – methods: numerical

1 INTRODUCTION

Mutual, catastrophic or non catastrophic, collisions among minor bodies of the Solar System are one of the most important processes influencing their evolution. Some fundamental properties of the Main Belt asteroids like the distribution of their sizes and spins or the existence and structure of asteroid families are the product of a complex interplay among collisions, gravitational and non gravitational processes. For this reason a deep knowledge of the statistics of impacts in those populations is a fundamental prerequisite for understanding their origin, evolution and present properties, as well as for the study of any process in the Solar System involving collisions like cratering, space weathering, surface refreshing, and so on.

Several works have been devoted to the statistics of collisions among minor bodies of the Solar System, and in particular among asteroids. The term “statistics of collisions” refers to the computation of impact rates and the derivation of distributions of any parameter related to the conditions at the moment of impact, mainly the impact velocity. The classical approach to tackle this problem is based on the pioneering works of [Öpik \(1951\)](#) and [Wetherill \(1967\)](#). Starting

from the same dynamical assumptions, [Kessler \(1981\)](#) developed an independent analytical formulation, confirming the results of [Wetherill \(1967\)](#). The approach of [Wetherill \(1967\)](#) was then refined by [Greenberg \(1982\)](#); [Bottke & Greenberg \(1993\)](#), and finally by [Bottke et al. \(1994\)](#), becoming the most widely used method of computation of the probability of impact among asteroids and of the distribution of their impact velocity. The method consists of an analytic description of the geometric condition of the crossing of asteroid orbits, and the final expression of impact rate is an integral to be computed numerically. The Öpik/Wetherill approach is based on some “canonical” assumptions about the dynamical evolution of the asteroid orbits. While the semimajor axes a , eccentricities e and inclinations I of the osculating orbits are supposed fixed, the rates of variation of the longitudes of the ascending nodes Ω and of the arguments of the pericenters ω are constant, or in other words the osculating orbits circulate uniformly. The motion of asteroids along their orbits is fully described by the Kepler’s second law. Such hypotheses are valid on first approximation for the large majority of Main Belt asteroids, and they are suitable for a good estimation of their mean lifetimes and impact velocity distributions.

A first attempt to overcome the limitations of the canonical conditions has been done by [Dell’Oro & Paolicchi](#)

* E-mail: delloro@arcetri.inaf.it

(1997). Later [Dell’Oro & Paolicchi \(1998\)](#) proposed a new method where the impact rate is expressed again in terms of an integral to be evaluated numerically, but the formula is based on a completely different and more general analytical approach with respect to [Bottke & Greenberg \(1993\)](#). Unlike the Öpik/Wetherill one, the [Dell’Oro & Paolicchi \(1998\)](#) approach does not need any *a priori* hypotheses about the angular elements Ω , ω and the mean anomaly M , and their method has been designed to be applied to cases characterized by any kind of distribution of Ω , ω and M , including the possibility of mutual correlation. The general formula for the computation of the impact rate requires as input (apart from the orbital elements a , e and I of both target and projectile) an analytic function of six variables describing the joint probability distribution of the elements Ω , ω and f (true anomaly) of target and projectile. In the framework of the mathematical formalism of [Dell’Oro & Paolicchi \(1998\)](#), the canonical conditions are re-formulated in terms of uniform distributions of node longitudes Ω , the pericenter arguments ω and mean anomalies M of the osculating orbits, and the assumption that no correlation exists among those three angles. The validity of the method of [Dell’Oro & Paolicchi \(1998\)](#) is even more general because it does not even need the hypothesis of constant a , e and I elements, which instead can be characterized by any statistical distribution. The only really necessary condition to use the method of [Dell’Oro & Paolicchi \(1998\)](#) is that none of the osculating elements a , e and I are correlated with any of the angular elements Ω , ω and M .

The method of [Dell’Oro & Paolicchi \(1998\)](#) has been used to study the statistics of collisions among different populations of minor bodies, with canonical and non canonical behavior, like for Main Belt asteroids taking into account the non uniform distribution of the lines of apses ([Dell’Oro et al. 1998](#)), or for bodies in resonant orbits like Trojans and Plutinos ([Dell’Oro et al. 1998, 2001, 2013](#)). It was also used to compute the rate of impacts among the members of asteroid families during the early stage of dynamical evolution of their orbit after the family formation ([Dell’Oro et al. 2002](#)). Concerning the problem of the role of the non uniform distribution of lines of apses of Main Belt asteroids we have to quote also the work of [JeongAhn & Malhotra \(2015\)](#) about the impact flux on planet Mars.

On the other hand, in the Solar System dynamical situations exist for which a correlation takes place between one (or more) of the elements a , e and I and one (or more) of the elements Ω , ω and M . Examples of this are orbits with large forced eccentricities and/or inclinations as expected by the Laplace-Lagrange linear secular theory or orbits in Kozai resonance ([Murray & Dermott 1999](#)). Some interesting efforts to compute the probability of collisions taking into account the effect of secular perturbations have been done by [Vokrouhlický et al. \(2012\)](#) and [Pokorný & Vokrouhlický \(2013\)](#), deriving analytical formulas in which the correlations between e - $\tilde{\omega}$ and I - Ω are taken into account. However [Pokorný & Vokrouhlický \(2013\)](#) theory is not fully general yet, requiring some limitations about the orbit of the target.

Alternative approaches are based on the numerical integration of the asteroids’ motion. If such computation is dense and extended enough, it is possible to register the temporal sequence of close approaches and extrapolate their rate to

the scale of mutual distances of the order of the physical size of the asteroids ([Marzari et al. 1996](#); [Dahlgren 1998](#)).

Another family of numerical techniques is based on a Monte Carlo approach, consisting in a random sampling of the phase space. The first serious implementation of a Monte Carlo approach was proposed by [Vedder \(1996\)](#), where it is shown how the rate of impacts can be expressed as a function of the parameters of a particular statistical distribution, that in turn can be estimated sampling the phase space randomly. This method was used with a certain success in case of collisions among the Main Belt asteroids ([Vedder 1998](#)), even if in some details the results differ from those obtained by [Bottke et al. \(1994\)](#). More recently [Rickman et al. \(2014\)](#) developed and tested two different methods of computation of the impact probability between a planet and a projectile with given semimajor axis, eccentricity and inclination, one similar to the “super-sizing” method introduced by [Horner & Jones \(2008\)](#) using the concept of Hill’s sphere, and the other one based on the computation of the minimum orbit intersection distance ([Wiżniowski & Rickman 2013](#)).

In this paper we present a new way to use the random samplings of the phase space in order to obtain the statistics of collisions among orbiting bodies. The method differs from both the approaches of [Vedder \(1996\)](#) and [Rickman et al. \(2014\)](#), even if it shares with them some fundamental ideas typical of any Monte Carlo strategy. The method is based on a new general and self-consistent mathematical formalism for the derivation of the temporal rate of close approaches between particles not interacting gravitationally with each other. This method does not assume any *a priori*, even if general, statistics of the minimum distances among the particles like in [Vedder \(1996\)](#), and it does not make use of the concept of the Hill’s sphere as in [Rickman et al. \(2014\)](#), revising the “super-sizing” approach from a more general point of view. Moreover, the final expressions that allow to derive the statistics of collisions from the series of samples of the phase space are extremely simple, and straightforwardly implementable. Originally this work had been triggered by the problem of computing impact rates among asteroids moving on orbits characterized by high values of forced eccentricity and inclination. Nevertheless this topic is only partially discussed in this paper. Here we prefer to focus on the fundamentals of the methodology and its validation, devoting future papers to special cases of scientific interest.

Sections 2 provides a general but self consistent description of the method, avoiding any technicality or mathematical proof of the formulas used in the text. A series of appendices are devoted to all technical details of the paper. In particular Appendix A, B, C and D contain the theoretical foundations of the method and the derivation of the formulas used in Section 2, while Appendix E focuses on the analysis and estimation of the numerical errors of the method. In Section 3 we show the practical use of the method for the important case of systems fulfilling the canonical Öpik/Wetherill dynamical assumptions, and the results are compared to those derived with independent methods. In Section 4 we show how to implement our method in order to take into account the secular perturbations. In particular, two examples with moderate and large forced eccentricities are discussed. Finally, Appendix F contains the results of further validation tests together with a list of revised values of the probabilities of collision and mean impact velocity for

selected targets that can be impacted by Main Belt asteroids.

2 DESCRIPTION OF THE ALGORITHM

The method presented in this paper belongs to the family of Monte Carlo approaches consisting in an exploration of the phase space of the system by means of a random sampling in order to extract information about the statistics of impacts of the particles composing the system. The main difference with other Monte Carlo approaches consists of not requiring any *a priori* choice of the variables to be sampled and the different way to extract from the list of the samples the statistical information about close encounters.

The fundamental prerequisite of the method is the availability of a suitable sampling model. For sampling model we mean any numerical algorithm able to generate a random series of dynamical configurations of the system and in particular the values of the relative distances and velocities among particles. The sampling model depends strictly on the dynamical properties of the system and it must be constructed from time to time accordingly. If only the overall statistical properties of the system are concerned, a sampling model is a sort of surrogate of a direct numerical integration able to simulate the detailed temporal evolution of the system. Taking into account a system composed of only two particles, its evolution is fully described by the two functions $\mathbf{r}_1(t)$ and $\mathbf{r}_2(t)$, where $\mathbf{r}_1$ and $\mathbf{r}_2$ are the position vectors of the two particles and t is the time. In principle, knowing $\mathbf{r}_1(t)$ and $\mathbf{r}_2(t)$, it is possible to count the number of times the two particles approach within a given distance R . So using an ideal numerical integrator able to simulate the evolution of the system for a sufficiently long time, it is possible to compute the mean number of close approaches per unit of time. How long the integration time should be it depends on the dynamical characteristics of the system. In any case, a reliable statistics of the close encounters of the particles requires a full simulation of all the possible kinematic configurations of the system. To do this, a sort of dynamical stability of the system is required. If the mechanical energy of our example system was positive the two particles would be unbound, and would move on hyperbolic orbits. Even if a close approach within a distance R is possible, it occurs only once and it is not possible to define any frequency of close encounters. The two particles should move inside a finite region of space from which they cannot exit, but this hypothesis alone would not be sufficient. We require that the probability distribution of all possible dynamical configurations of the system does not depend on time. In this context, for dynamical configuration of our system we intend positions $\mathbf{r}_1$, $\mathbf{r}_2$ and velocities $\mathbf{v}_1 = \dot{\mathbf{r}}_1$, $\mathbf{v}_2 = \dot{\mathbf{r}}_2$ of the two particles at a given time t . Therefore, if $\psi(\mathbf{r}_1, \mathbf{v}_1, \mathbf{r}_2, \mathbf{v}_2)d\Gamma$ is the probability to find the two particles with positions and velocities inside the infinitesimal volume $d\Gamma$ of the phase space around the values $(\mathbf{r}_1, \mathbf{v}_1, \mathbf{r}_2, \mathbf{v}_2)$, we assume that the probability density ψ does not depend on time. A sampling model is a numerical algorithm able to generate a list of values of $(\mathbf{r}_1, \mathbf{v}_1, \mathbf{r}_2, \mathbf{v}_2)$ the parent distribution of which is ψ . The temporal stability of the function ψ is the basic condition of validity of all the analytical methods developed to compute the probability of collisions among minor bodies,

although not explicitly given in these terms. The dynamical stability of the system is rather expressed using other variables different from positions and velocities. Indeed, the development of an analytical method as well as the construction of a sampling model rely on the availability of a dynamical model of evolution of the system describing its essential features. Examples of sampling models will be further described when practical cases will be discussed.

The random sampling of the phase space is the essential part of any Monte Carlo approach developed so far. What makes the methodology presented in this work different from the previous approaches is the way to use the samplings to compute the frequency of close encounters. It is important to stress that in this work we do not take into account the case of interacting bodies, or in other words hereinafter we assume that either during the close encounters or in any other phase of their evolution the particles do not attract each other. So no gravitational focusing is included in the computations. The method can be used for the study of the collisions among minor bodies the masses of which are negligible, but probably it could have more or less limitations if used for the study of the statistics of close encounters and collisions between minor bodies and planets.

A close encounter within a distance R occurs when the relative distance $r(t) = |\mathbf{r}_2(t) - \mathbf{r}_1(t)|$ between the two particles has a local minimum r_m and $r_m < R$. Arbitrarily we name “target” one of the particles and “impactor” the other one, and we chose a reference system such that the target is at rest in the origin. In this reference system, around the close encounter epoch, the impactor spends a time Δt inside a sphere of radius R centered around the target. We name Δt the duration of the close encounter. Following the hypothesis of non interacting particles, we assume that during the interval Δt the *relative* motion of the impactor can be approximated as a rectilinear trajectory with constant velocity. Such hypothesis is not true in general, but we expect that it is better and better fulfilled for smaller and smaller values of R . It is important to stress that in this work we are not interested in evaluating the frequency of any kind of close encounter in general, but only of those occurring at a distance comparable to the physical dimensions of involved bodies. If $\phi(R)$ is the mean number per unit time of close encounters within a distance R , our goal is to study the values of $\phi(R)$ for $R \rightarrow 0$, ignoring the cases with large R . In any case the function $\phi(R)$ is *a priori* computed for a wide range of values of R always assuming a rectilinear relative motion, without proving the accuracy of this assumption, and then the relevant information is extracted *a posteriori* from the trend of $\phi(R)$, as explained later on.

The last assumption of the algorithm is about of the distribution of relative minimum distances r_m of close encounters within R . Each time the impactor crosses the sphere of radius R , the minimum approach distance is different. We assume that the probability that the minimum approach distance is less than r_m is proportional to r_m^2 . This follows from a pure geometric cross-section description of the close encounter probability. Also this hypothesis is not true for all values of R , and in particular for large R , while we expect that it is satisfactorily fulfilled for small R .

The two hypotheses about the motion of the impactor inside the sphere of radius R are used to compute the mean duration $\overline{\Delta t}$ of a close encounter with a given value of the rel-

ative velocity $v = |\mathbf{v}_2 - \mathbf{v}_1|$, that we assume constant at least during the crossing of the sphere. As shown in Appendix A, the average value of the transit duration is $\overline{\Delta t} = (4/3)(R/v)$.

The importance of the parameter $\overline{\Delta t}$ is due to its connection with the probability to find the two particles at a distance smaller than R and the temporal rate of the close encounters. In fact if we followed the evolution of the system for a period of time T and we counted the number of close encounters $N(R)$ within a distance R that occurred meanwhile, by definition the mean number of event per unit time would be $\nu(R) = N(R)/T$. On the other hand we may also write $\nu(R) = p(R)/\tau(R)$, where $p(R) = S(R)/T$, $\tau(R) = S(R)/N(R)$ and $S(R)$ is the sum of the durations Δt of the $N(R)$ close encounters. Clearly, $p(R)$ is the probability to find the two particles at a distance smaller than R *observing them at any instant of time randomly chosen*, while $\tau(R)$ is the average duration of the close encounters within a distance R , or in other words $\tau(R)$ is the average of $\overline{\Delta t}$ taking into account the distribution of values of v .

The probability $p(R)$ is the typical quantity that can be evaluated by means of a sampling model. If we generate, using the sampling model, a sufficiently large number n of dynamical configurations, we can compute for each of them the relative distance r_k and the relative velocity v_k of the two particles, where k is the index of the k -th generated configuration, running between 1 and n . If $n(R)$ is the number of configurations such that $r_k < R$, then $p(R) = n(R)/n$. As shown in Appendix B, the temporal rate of the close encounters within a distance R can be computed simply as:

$$\phi(R) = \frac{3}{4} \frac{1}{R} \frac{1}{n} \sum_{k:r_k < R} v_k \quad (1)$$

where the sum includes *only* the $n(R)$ sample relative velocities v_k for which the corresponding sample relative distances r_k are smaller than R , while n is the *total* number of sample configurations (r_k, v_k) . As it results from Eq. (1), each sample configuration contributes to the temporal rate with a different term proportional to v_k . This is in accordance with the improvement of the method of Öpik/Wetherill introduced by Bottke et al. (1994), where each different orbital geometry is properly weighted according to its specific probability. In the framework of our method and formalism this weight appears as the value of the relative velocity.

The statistics of the close encounters does not regard only the frequency of the events, but also the distribution of any parameter related to the close encounter. From this point of view, we assume that the value of the parameter does not change during the close encounter, i.e. on a timescale corresponding to the duration of the close encounter. As for the relative velocity, this approximation is expected to be better and better fulfilled as R is smaller and smaller. The parameter under investigation q may be a quantity that can be computed from positions and velocities of the particles, like the relative velocity itself and its orientation in space, or it is an intermediate quantity for the computation of positions and velocities, like the parameters describing the orientation of the orbit or the location along the orbit. The sampling model should provide together with the sample relative distance and velocity (r_k, v_k) also the corresponding sample value q_k of the parameter. In Appendix C it is shown that the fraction of the close encounters

occurred per unit of time within a distance R such that the value of the parameter belongs to an interval of values $\mathcal{Q}$ is:

$$\Psi(R, \mathcal{Q}) = \frac{\sum_{k:r_k < R, q_k \in \mathcal{Q}} v_k}{\sum_{k:r_k < R} v_k} \quad (2)$$

where the sum in denominator includes all the sample velocities for which $r_k < R$, while the sum in numerator includes *only* the sample velocities such that $r_k < R$ and $q_k \in \mathcal{Q}$. Moreover, the expected value of the parameter q is given by:

$$\langle q(R) \rangle = \frac{\sum_{k:r_k < R} q_k v_k}{\sum_{k:r_k < R} v_k} \quad (3)$$

where the symbol $\langle q(R) \rangle$ emphasizes that the mean value of q refers to only the close encounters within a distance R , and both sums on the numerator and denominator include all the sample velocities v_k such that $r_k < R$, while q_k is the corresponding value of the parameter for each pair (see Appendix C for details). In few words, the expected value of a parameter q for the close approaches within a distance R is simply the weighted average of the sample values q_k by the corresponding relative velocity. One of the most important close encounter parameter is the relative velocity itself, the mean value of which is then correctly evaluated by:

$$\langle v(R) \rangle = \frac{\sum_{k:r_k < R} v_k^2}{\sum_{k:r_k < R} v_k} \quad (4)$$

Equations (1)-(4) are the core tools of our algorithm, and they show in a very simple way how to obtain the statistics of close encounters within a given distance R from the series of samples (r_k, v_k, q_k) . Those equations are the same whatever sampling model is used, because of it is not important which procedure is followed to generate the series of random samples, provided that their distributions matches the density distribution ψ of the dynamical configuration of the system.

The computation of the quantities $\phi(R)$, $\Psi(R, \mathcal{Q})$ and $\langle q(R) \rangle$ by means of Eqs. (1)-(3) is not a deterministic process being based on a Monte Carlo sampling. All the sums $(\sum_{k:r_k < R} v_k)$, $(\sum_{k:r_k < R, q_k \in \mathcal{Q}} v_k)$ and $(\sum_{k:r_k < R} q_k v_k)$ are random variates the values of which depends on the particular random generation sequence of the samplings (r_k, v_k, q_k) . Repeating the computation with new independent series of samplings, the values of the quantities provided by Eqs. (1)-(3) fluctuate more or less around mean values. Only for $n \rightarrow \infty$ the formulas converge to the exact theoretical values. Both the numbers of terms of those sums and the values of each term are random variables. In particular in sums $(\sum_{k:r_k < R} v_k)$ and $(\sum_{k:r_k < R} q_k v_k)$ the number of terms is $n(R)$, the number of samples for which the relative distance r_k is less than R . Typically $n(R)$ is a Poisson deviate, and it can be null if no dynamical configuration is yet found with the required property.

If $\bar{v}$ is the mean value of the relative velocity generated by the sampling model, $\sum_{k:r_k < R} v_k \sim n(R)\bar{v}$. But $n(R)$ is a random variable with mean and variance equal to $np(R)$. Moreover, we expect that, for R small enough, the probability $p(R)$ to find the two particles at distance each other shorter than R is proportional to R^3 . So, in conclusion, the standard deviation of $\phi(R)$, computed starting from different and independent series of samplings, is $\sigma_\phi \propto R^{1/2}n^{-1/2}$. It is necessary to increase by a factor 100 the total number of samples (that is the computational time) in order to reduce by a factor 10 the error of $\phi(R)$. This is a common drawback of every Monte Carlo approach, consisting of an intrinsic numerical inefficiency.

On the other hand an important advantage of this Monte Carlo approach is that equations (1)-(4) can be used even in the case that the system is composed of more than two particles. Let us imagine that our system is composed of N different particles. For each pair of particles, identified by the indices i and j (with $i \neq j$), we can compute separately the rate of close encounters $\phi_{ij}(R)$, the distribution $\Psi_{ij}(R, \mathcal{Q})$ of the parameter q and its expected value $\langle q(R) \rangle_{ij}$, using separated series of samplings $(r_{ijk}, v_{ijk}, q_{ijk})$. On the other hand we may be interested in the averages of the above quantities over the ensemble of all possible pairs of particles, respectively given by:

$$\overline{\phi(R)} = \frac{\sum_{i \neq j} \phi_{ij}(R)}{M} \quad (5)$$

$$\overline{\Psi(R, \mathcal{Q})} = \frac{\sum_{i \neq j} \Psi_{ij}(R, \mathcal{Q})}{M} \quad (6)$$

$$\overline{\langle q(R) \rangle} = \frac{\sum_{i \neq j} \langle q(R) \rangle_{ij}}{M} \quad (7)$$

where $M = N(N-1)$ is the number of all possible pairs. The same averages are defined in the case that we are interested in the statistics of close encounters between a fixed particle (target) and a swarm of other N particles (impactors), provided that $M = N$, that is the number of all the possible pairs target-impactor. In any case, as it is shown in Appendix D, all the three averaged quantities $\overline{\phi(R)}$, $\overline{\Psi(R, \mathcal{Q})}$ and $\overline{\langle q(R) \rangle}$ can be evaluate directly by Eqs. (1)-(4) using a *unique* series of random samplings (r_k, v_k, q_k) , provided that at each new generation of a system configuration the two particles are *randomly* chosen among those constituting the system. The basic requirement is that at each generation the probability to draw lots a particular couple of particles is the same for all possible couples. The major advantage of this approach is numerical. As explained before, the numerical uncertainty in computing Eqs. (1)-(4) comes from the random fluctuations inherent in the random generation of the samples (r_k, v_k, q_k) . The variance of such fluctuations scales as $1/n$, where n is the total number of samples. If we devote to *each* pair of particles a number of n samples, the variance of the average rate $\overline{\phi(R)}$ scales as $1/(nM)$, where M is the number of terms in the average (the number of possible pairs). This means that if we do not need to know the single $\phi_{ij}(R)$ with a large precision we can reduce the number of samples n devoted to each pair by a factor M in order to curb the total number of samples generated to compute

$\overline{\phi(R)}$. The advantage is that, given the level of variance we need for $\overline{\phi(R)}$, the computational time does not depend on the number of particles composing the system. The extreme case consists of systems composed of a continuous distribution of orbits (particles), the orbital elements of which are generated directly by the sampling model (see an example in the Section 3).

The last step of our method consists in extrapolating the parameters of the statistics for $R \rightarrow 0$. All quantities $\phi(R)$, $\Psi(R, \mathcal{Q})$ and $\langle q(R) \rangle$, either referring to the case of a single pair of particles or averaged over different couples of particles, are all functions of the close encounter range R . As pointed out before, the only interval of values of R we are interested in is that corresponding to the physical size of the particles under investigation. On the other hand, it is almost never possible to compute $\phi(R)$, and the other related quantities, directly for values of R small enough. The precision in the computation of $\phi(R)$ depends on the number $n(R)$ of sample configurations with $r_k < R$ that we have generated. The larger $n(R)$ is the smaller the uncertainty of $\phi(R)$ is. But the probability $p(R)$ to generate such sample configurations decreases very fast with R and typically $p(R) \propto R^3$, therefore the computational time needed to generate the same number $n(R)$ of sample configurations for different values of R is proportional to R^{-3} . For example, in the case of asteroids we would like compute $\phi(R)$ for values of R of the order of one to hundred kilometers. As we see later, for Main Belt asteroids with typical size of the orbits of the order of $2 \div 3$ AU the probability to generate sample configurations with relative distance less than 1-100 km is so small as to requires a prohibitive computational effort. On the other hand we have some conceivable expectations about the trend of $\phi(R)$ for $R \rightarrow 0$. Ignoring any mutual gravitational effect, $\phi(R)$ should be proportional to R^2 for R small enough, a trend simply coming from the pure geometrical cross-section of the close encounter. On the other hand such expectation is the basis of the definition of *intrinsic* probability of collision P_i introduced by Wetherill (1967), and widely used in the study of statistics of collisions among asteroids. The quantity P_i is defined with respect a pair of bodies with intersecting orbits, and it depends only on the orbital parameters regardless the physical sizes of the two bodies. It is defined as the mean number of close approaches per unit of time within a distance of 1 km, and it is usually expressed in units of $\text{km}^{-2} \text{yr}^{-1}$. In other term $P_i = \phi(R)$ for $R = 1$ km. Following Wetherill (1967), the probability of collision, during an interval of time T , between two bodies with radii R_t and R_p , moving on the orbits for which P_i is known, is given by $(R_t + R_p)^2 P_i T$, if the radii are expressed in kilometers and T in years. However we have to note that, from a strictly mathematical point of view, the assumption that $\phi(R) \propto R^2$ may not be true in some special cases. In the case of two orbits always perfectly coplanar, for example if both their inclinations are null, we lose the three-dimensionality of the system, the rate of close approaches cannot be proportional to the usual geometrical cross section but it results to be proportional to R , for all values of R . Apart from such special case, the typical situation is that $\phi(R)$ is proportional to R for large values of R , and $\phi(R) \propto R^2$ for small R . The exact meaning of “large” and “small” depends on the geometry of the orbits and in particular on their inclinations. Let us take into ac-

count orbits with typical semimajor axis a , eccentricity e and inclination I . The maximum distance from the ecliptic plane that the particle can reach is typically about $a(1+e)\sin I$. For values of R much larger than $a(1+e)\sin I$ the system appears almost bi-dimensional and so $\phi(R) \propto R$, but for $R \ll a(1+e)\sin I$ the three-dimensionality of the system is fully recovered and $\phi(R) \propto R^2$. The threshold dividing the two regimes depends on I . In any case, the general strategy is to search for a value R_ϕ such that for $R < R_\phi$ the rate $\phi(R) \propto R^2$. In an equivalent manner we look for a value R_ϕ such that for $R < R_\phi$ it results that $\phi(R)/R^2$ is constant. In such interval of values of R the ratio $\phi(R)/R^2$ is by definition equal to P_i . For what concerns the distribution and the mean value of a close encounter parameter q computed by means of Eqs. (2) and (3) we expect that they tend asymptotically to a finite value for $R \rightarrow 0$. In other terms we look for a value R_q such that for $R < R_q$ the mean value $\langle q(R) \rangle$ is constant. It is worthwhile to note that in general the two thresholds R_ϕ and R_q , if they exist, could not be the same. More in details, for what concerns the mean value of the close encounter relative velocity $\langle v(R) \rangle$, its limit U_m for $R \rightarrow 0$ is our estimation of the mean impact velocity.

In conclusion, our algorithm can be summarized as it follows. A series of values R_i of the close encounter range is defined. Generally the values R_i are logarithmically scaled covering a wide interval of orders of magnitude of the relative distance between the particles. The computation consists in a recursive loop. Each time the loop restarts, a new dynamical configuration is generated. If the index k identifies the k -th configuration, for each value R_i the sums $S_{i,v} = (\sum_{k:r_k < R_i} v_k)$ and $S_{i,v^2} = (\sum_{k:r_k < R_i} v_k^2)$ are defined. Moreover, if we are interested in the statistics of a particular parameter q , the additional sums $S_{i,\mathcal{Q}_j} = (\sum_{k:r_k < R_i, q_k \in \mathcal{Q}_j} v_k)$ and $S_{i,qv} = (\sum_{k:r_k < R_i} q_k v_k)$ are defined, where $\mathcal{Q}_j$ is a suitable set of bins of values of the parameter q used to build up the distribution of q . Anyway all sums S are initially set equal to zero. The loop consists of the following four steps:

- (L1) a random dynamical configuration is generated using the sampling model. The result consists in the values of the relative distance r_k , the corresponding value of the relative velocity v_k , and the value q_k of the parameter we are interested in. The sampling model, or in other words the algorithm used to generate r_k , v_k and q_k depends on the particular dynamical system under investigation and it is developed apart;
- (L2) the total number of generated configurations n is incremented by one, and the values of the sums $S_{i,v}$, S_{i,v^2} and $S_{i,qv}$ are incremented respectively by v_k , v_k^2 and $q_k v_k$ if $r_k < R_i$, while the sums $S_{i,\mathcal{Q}_j}$ are incremented by v_k if $r_k < R_i$ and $q_k \in \mathcal{Q}_j$;
- (L3) the values of $\phi(R_i)$, $\Psi(R_i, \mathcal{Q}_j)$, $\langle q(R_i) \rangle$ and $\langle v(R_i) \rangle$ are re-computed on the basis of Eqs. (1)-(4) using the new values of the sums S . Moreover, the errors in evaluating $\phi(R)$, $\langle q(R) \rangle$ and $\langle v(R) \rangle$ due to the random fluctuations in the generation of r_k , v_k and q_k are estimated (see Appendix E for details);
- (L4) the series of values $\phi(R_i)$, $\langle q(R_i) \rangle$ and $\langle v(R_i) \rangle$ are analyzed to identify, if it exists, a value R_ϕ such that for all $R_i < R_\phi$ it results that $\phi(R_i) \propto R_i^2$, or in other terms if $\phi(R_i)/R_i^2$ are compatible with a constant value for $R_i < R_\phi$ taken into account the random fluctuations. A similar check

is done also for $\langle v(R_i) \rangle$ and $\langle q(R_i) \rangle$, looking for values R_v and R_q such that $\langle v(R_i) \rangle$ and $\langle q(R_i) \rangle$ become constant respectively for $R < R_v$ and $R < R_q$, compatibly with the estimated random fluctuations. If the result of these checks is positive, the loop is interrupted, otherwise it restarts from the step (L1).

The final result of the processing is the estimation of the asymptotic value of the ratio $\phi(R)/R^2$ for $R \rightarrow 0$, as our evaluation of P_i , and the limit values of $\langle v(R) \rangle$ and $\langle q(R) \rangle$, respectively as estimation of U_m and of the mean of q at impact. The algorithm fails when the expected limit trends are not identified with acceptable accuracy. This can happen if the total number of samplings n is not sufficient, and the algorithm requires a much larger number of samplings with consequently much long computational time. In some cases, depending on the properties of the dynamical system, the necessary computing time turns out to be prohibitive.

3 CANONICAL STATISTICS

The first example of use of the method described in Section 2 we want to discuss in this paper regards the statistics of collisions among asteroids assuming the same dynamical hypotheses the Öpik/Wetherill theory is based on (Öpik 1951; Wetherill 1967). Such assumptions are:

- (i) the semimajor axes a , eccentricities e and inclinations I of the osculating orbits are fixed;
- (ii) the rate of variation of the longitudes of the nodes Ω and arguments of the pericenters ω of the osculating orbits are constant, or in other words the osculating orbits circulate uniformly;
- (iii) the motion of the asteroids along their osculating orbits is fully described by the Kepler’s second law;

The condition (iii) is nothing else to state that the mean anomalies M of the bodies uniformly change with time. Therefore the conditions (ii) and (iii) entail that all the possible values between 0 and 2π of the three angles Ω , ω and M are equally probable, if we observe the bodies at any instant of time randomly chosen. Moreover neither no correlation exists among the three angles of the same orbit nor between the angles of different orbits. Let us consider a system composed of only two particles, identified by the indices i and j . The values of the orbital elements (a_i, e_i, I_i) and (a_j, e_j, I_j) are fixed once and for all. The sampling model correctly describing the dynamical properties of the system consists of the following steps:

- (C1) the values of the angles Ω_i , ω_i , M_i of the i -th particle, and the angles Ω_j , ω_j , M_j of the j -th particle are randomly and independently chosen in the interval $[0, 2\pi]$;
- (C2) the vector position $\mathbf{r}_i$ and vector velocity $\mathbf{v}_i$ of the i -th particle are computed from the orbital elements a_i , e_i , I_i , Ω_i , ω_i and M_i . The vector position $\mathbf{r}_j$ and vector velocity $\mathbf{v}_j$ of the j -th particle are computed from the orbital elements a_j , e_j , I_j , Ω_j , ω_j and M_j ;
- (C3) the relative distance $r = |\mathbf{r}_i - \mathbf{r}_j|$ and the relative velocity $v = |\mathbf{v}_i - \mathbf{v}_j|$ are computed;

The steps (C1)-(C3) are the procedure to be used in the step (L1) of the algorithm loop of Section 2, if canonical

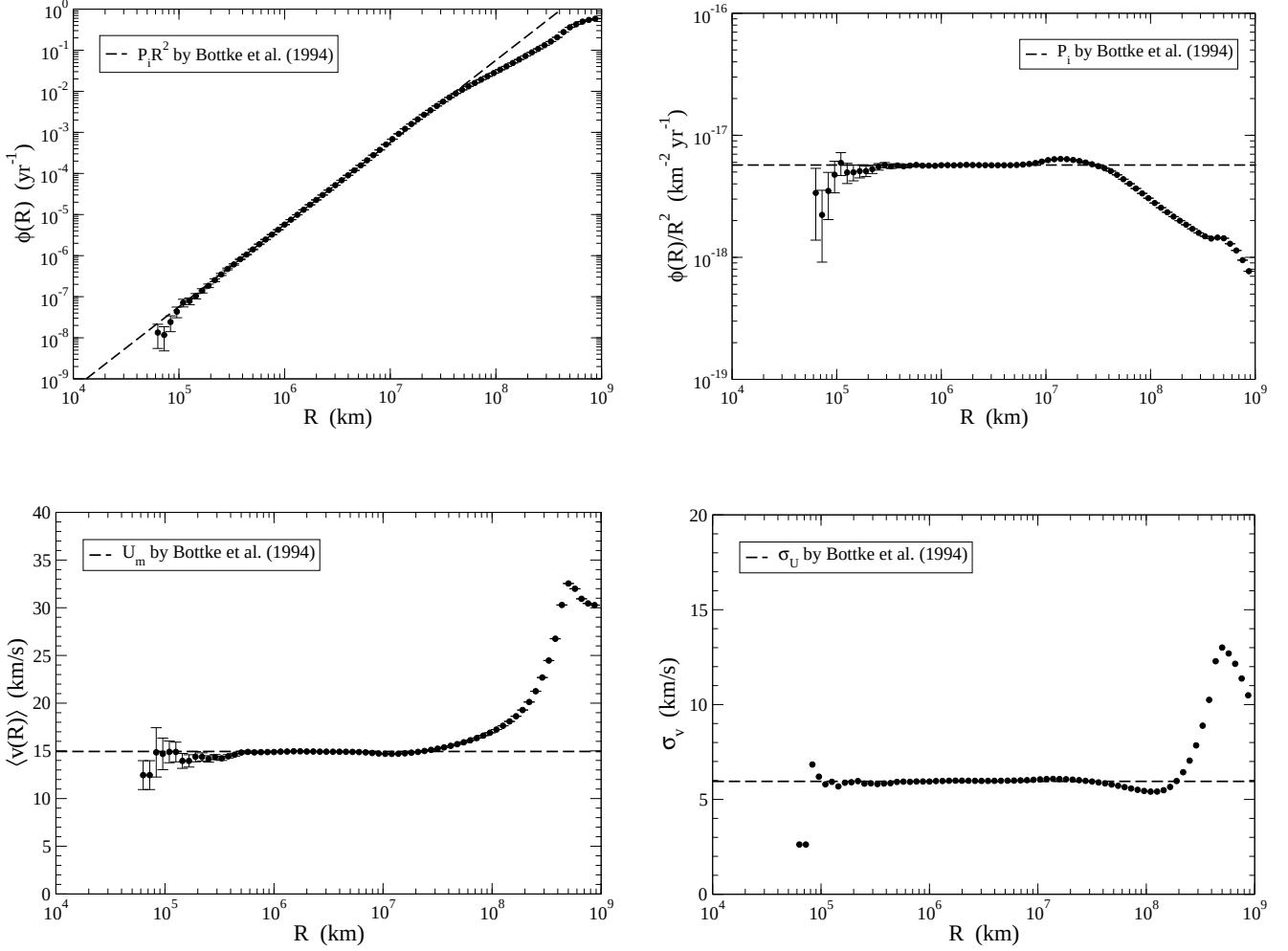


Figure 1. Canonical statistics of close encounters between two orbits with elements $a = 3.42$ AU, $e = 0.578$, $I = 0.435$ rad, and $a = 1.59$ AU, $e = 0.056$, $I = 0.466$ rad. Top left: frequency of close encounters $\phi(R)$ computed using Eq. (1) versus R . Top right: ratio $\phi(R)/R^2$ versus R . Bottom left: mean close encounter velocity $\langle v(R) \rangle$ computed using Eq. (4) versus R . Bottom right: standard deviation of the close encounter velocity versus R . The long-dashed lines represent the intrinsic probability of collision P_i , the mean impact velocity U_m and the standard deviation of the impact velocity σ_U computed using the method of Bottke et al. (1994).

conditions (i)-(iii) are assumed. In the case we deal with a group of asteroids composed of more than two bodies, a step (C0) must be included just before the step (C1), in which two asteroids are randomly selected provided only they are distinct ($i \neq j$). In this case, each new generation of the values r and v starts from new sets of the orbital elements (a_i, e_i, I_i) and (a_j, e_j, I_j) .

Canonical statistics of collisions among asteroids is an already deeply investigated scientific case, and well established computational tools have been developed in particular by Greenberg (1982) and Bottke et al. (1994). The more general approach of Dell’Oro & Paolicchi (1998) proved to be equivalent, if canonical hypotheses are assumed, to Greenberg (1982) and Bottke et al. (1994), providing the same results for the same test cases. In Appendix F a series of validation tests of our Monte Carlo algorithm are described, comparing the results of this method with those provided by Bottke et al. (1994) and Dell’Oro & Paolicchi

(1998). Here we discuss only some example cases in order to show the use of our method.

The first example regards a pair of bodies with orbital elements $a = 3.42$ AU, $e = 0.578$, $I = 0.435$ rad, and $a = 1.59$ AU, $e = 0.056$, $I = 0.466$ rad. This case has been already studied by Bottke et al. (1994) and Dell’Oro & Paolicchi (1998). The intrinsic probability of collision computed using the methods of Bottke et al. (1994) and Dell’Oro & Paolicchi (1998) results to be $P_i = 5.70 \times 10^{-18} \text{ km}^{-2} \text{ yr}^{-1}$, while the mean impact velocity and the standard deviation of the impact velocity are respectively $U_m = 14.94 \text{ km/s}$ and $\sigma_U = 5.95 \text{ km/s}$. We computed the quantities $\phi(R)$ and $\langle v(R) \rangle$ by means of Eqs. (1) and (4) using a number of samplings $n = 10^{12}$. The results are shown in Fig. 1. A well identified trend $\phi(R) \propto R^2$ holds for values of R below 10^7 km , about 4 % of the semimajor axis of the smaller orbit. Also $\langle v(R) \rangle$ and the standard deviation of the relative velocity $\sigma_v(R)$ become constant for $R < 10^7 \text{ km}$.

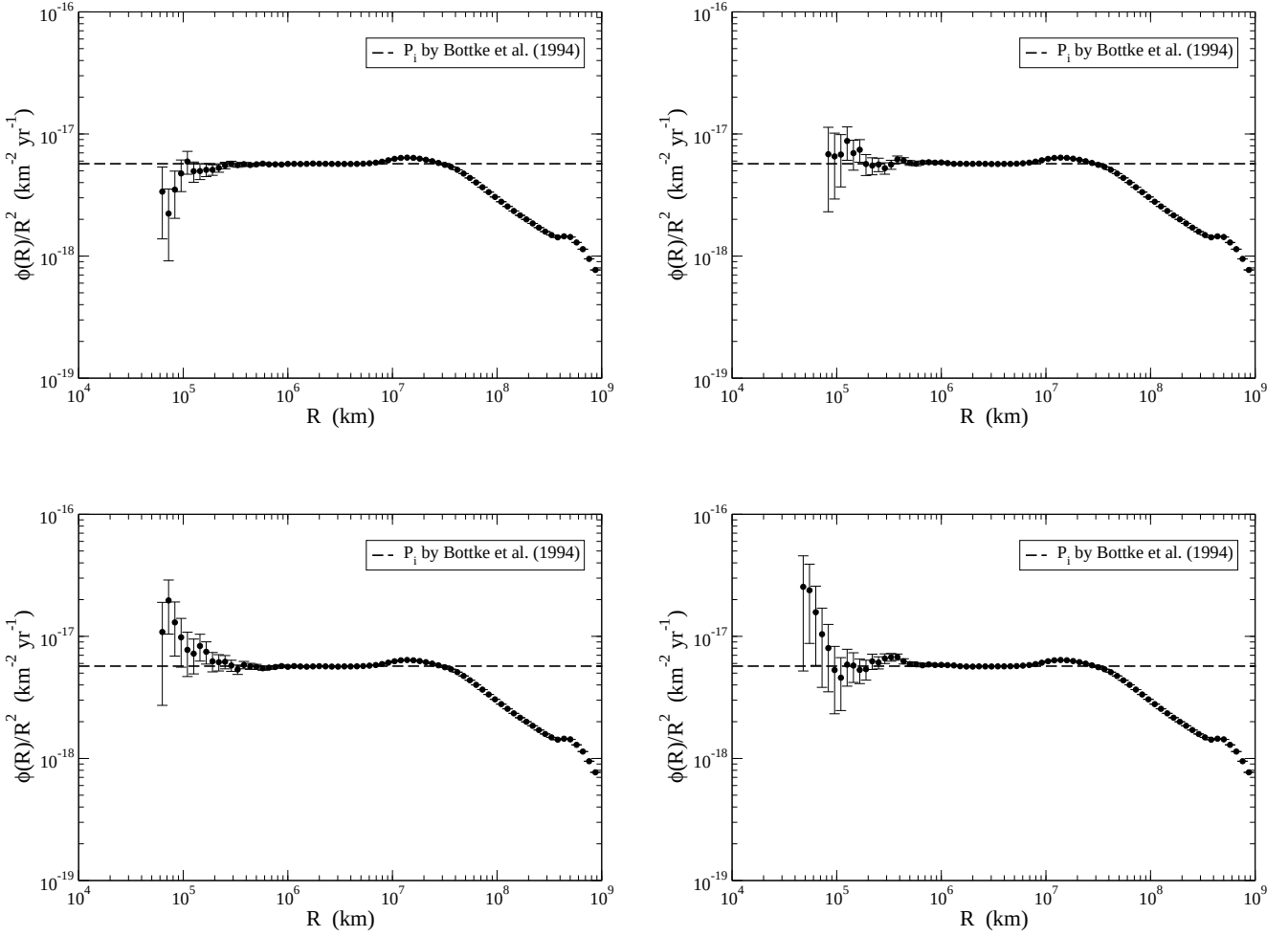


Figure 2. The plot $\phi(R)/R^2$ versus R of Fig. 1 at top right (top left in this figure) is compared with three other similar plots obtained with different random sampling series (top left, bottom left and right).

The standard deviation of the frequency distribution of the relative velocity for the close encounter within a distance R is computed as $\sigma_v^2(R) = \langle v^2(R) \rangle - \langle v(R) \rangle^2$, where, according to Eq. (3), the expected value of the square of the relative velocity is:

$$\langle v^2(R) \rangle = \frac{\sum_{k:r_k < R} v_k^3}{\sum_{k:r_k < R} v_k} \quad (8)$$

The asymptotic values of $\phi(R)/R^2$, $\langle v(R) \rangle$ and $\sigma_v(R)$ for $R \rightarrow 0$ are fully compatible with the parameters P_i , U_m and σ_U computed independently, and represented by long-dashed lines in the plots.

Below 10^5 km the random fluctuations of this particular sampling become significant. The length of error bars is equal to the expected standard deviation of the vertical shift of the respective points from their average positions, that we would obtain with an infinite number of samples or repeating the sampling a large number of times. Fig. 2 gives an idea of the extent of the random fluctuations of the

value of $\phi(R)/R^2$ computed using Eq. (1). The plots show $\phi(R)/R^2$ versus R obtained using four sample series (r_k, v_k) independently generated but with the same number of samples. It is important to stress that the fluctuations of the values $\phi(R)$, $\langle v(R) \rangle$ and $\sigma_v(R)$ for different values of R are not independent but correlated with each other. In fact, the parameter $\phi(R)$ is obtained using the values of v_k such that $r_k < R$, that are also included among the samples used to compute $\phi(R')$ with $R' > R$. In this way systematic trends in fluctuations of the values $\phi(R)$ can take place (like that shown in Fig. 2 bottom right), but they have nothing to do with the mean trend of the function $\phi(R)$ versus R .

Fig. 3 shows some frequency distributions of the relative velocity computed using Eq. (2). In this case the parameter q is the relative velocity itself. The interval of values of v from 0 km/s to 30 km/s has been divided into one thousand bins $\mathcal{V}_i = [v_i, v_i + \Delta v]$, each $\Delta v = 30$ m/s wide. So from Eq.

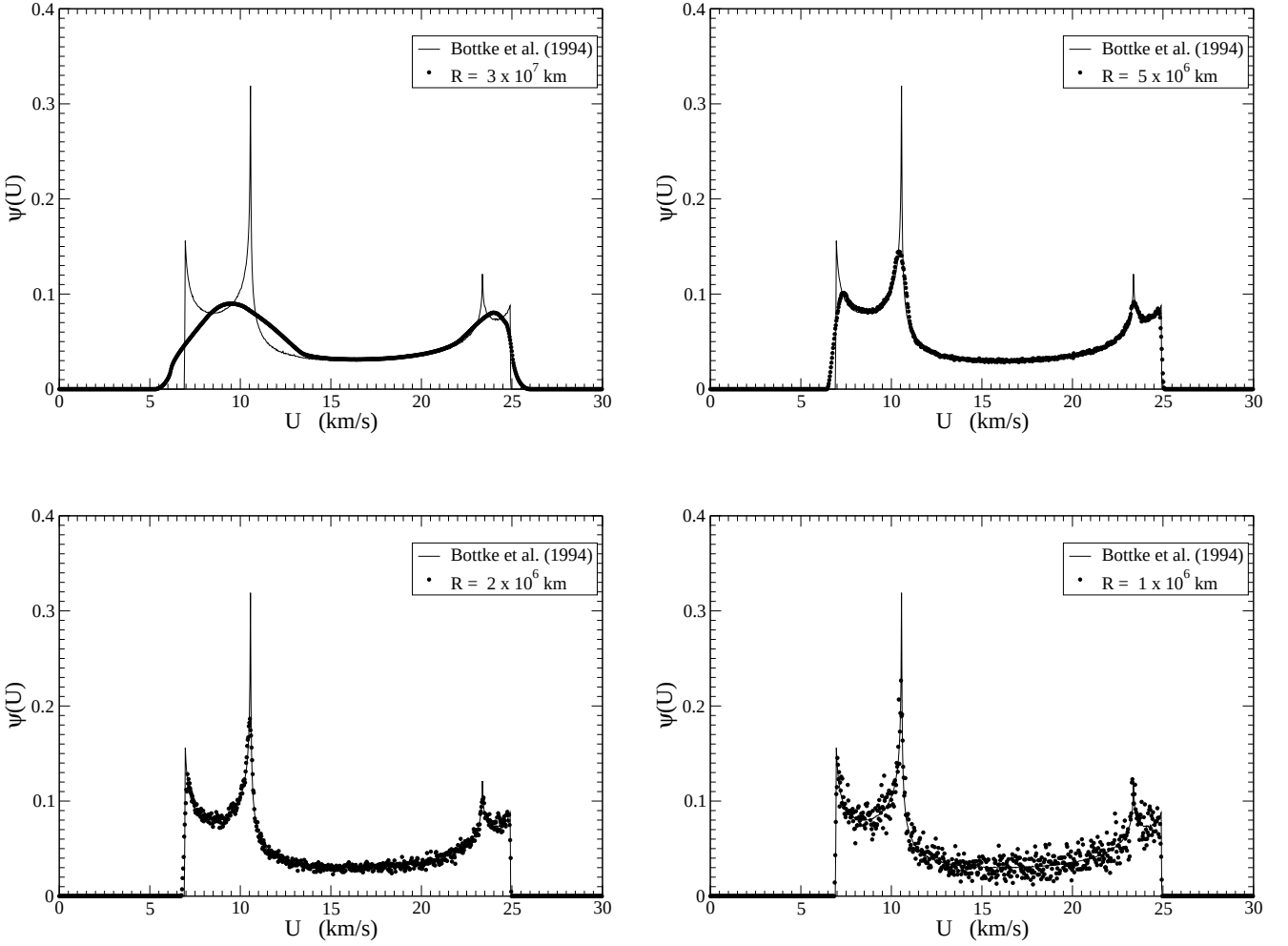


Figure 3. For the same case of Fig. 1, distributions of the close encounter velocity (dots) obtained using Eq. (2) for the close encounters occurring within different distance limits R (3×10^7 km at top left, 5×10^6 km at top right, 2×10^6 km at bottom left and 10^6 km at bottom right). A set of bins 30 m/s wide are used as intervals $\mathcal{Q}$ in Eq. (2). Each plot is compared with the impact velocity distribution obtained by Bottke et al. (1994) (thin line).

(2) the fraction of events occurring with relative velocity belonging to each bin is:

$$\Psi(R, \mathcal{V}_i) = \frac{\sum_{k:r_k < R, v_k \in [v_i, v_i + \Delta v]} v_k}{\sum_{k:r_k < R} v_k} \quad (9)$$

In plots in Fig. 3 the values $\Psi(R, \mathcal{V}_i)/\Delta v$ versus v_i are represented as black dots. Four different cases are taken into account corresponding to the distribution of the relative velocities for the close encounter occurring within distances $R = 3 \times 10^7$ km, $R = 5 \times 10^6$ km, $R = 2 \times 10^6$ km and $R = 1 \times 10^6$ km. The plots show how, decreasing R , the distributions $\Psi(R, \mathcal{V}_i)$ converge to the distribution of the impact velocity predicted by Bottke et al. (1994) and drawn in each plot as a thin line. Obviously, the random noise increases for R smaller and smaller. The total number of samples (r_k, v_k) used in Eq. (9) was $n = 10^{12}$. The probability to

generate a dynamical configuration with $r_k < R = 10^6$ km resulted to about $p(R) = 1.9 \times 10^{-8}$, and the corresponding number of samples was about 1.9×10^4 . So in mean about twenty samples fall into each of the one thousand velocity bins $\mathcal{V}_i$, with a standard deviation of the order of 20 %. On the other hand, all the features of the impact velocity distribution are correctly reproduced on average, including the peaks corresponding to the encounter close to the pericenters and apocenters of the two orbits.

The second example regards two orbits with orbital elements much more similar than in previous case in order to highlight the expected behavior for $R \rightarrow 0$, in particular for what concerns $\phi(R)$. The two orbits have orbital elements $a_0 = 2.5$ AU, $e_0 = 0$, $I_0 = 0$ deg, and $a_1 = 2.5$, $e = 0.001$, $I = 0.05$ deg. The smaller differences between the orbital elements allow to explore values of R much smaller than in the previous example without an excessive computational effort. Fig. 4 shows the result of this exercise. The methods of Bottke et al. (1994) and Dell’Oro & Paolicchi (1998) give

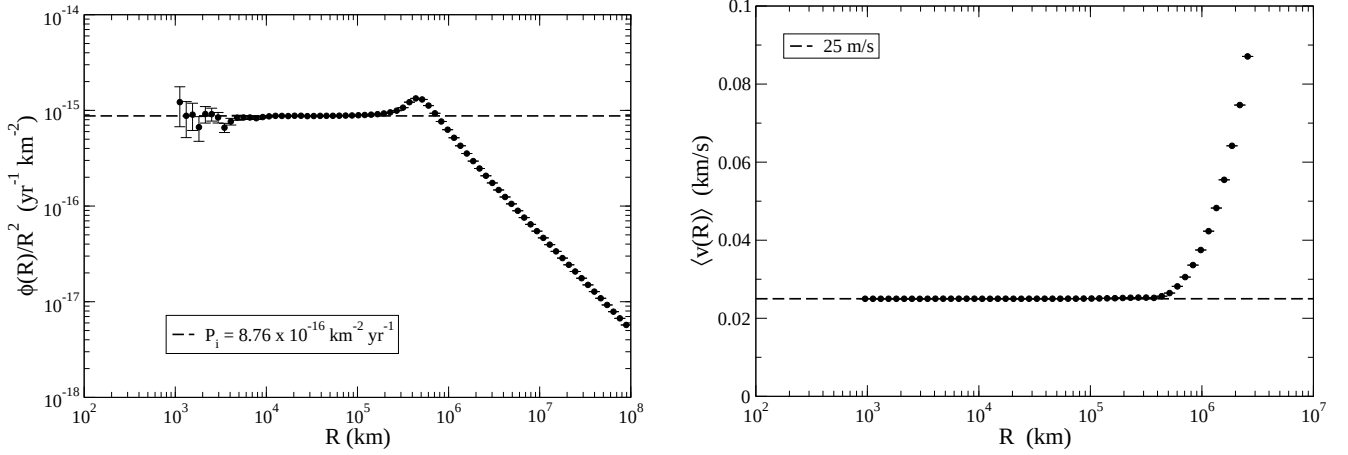


Figure 4. Close encounters between two orbits with elements $a = 2.5$ AU, $e = 0$ and $I = 0$ deg, and $a = 2.5$ AU, $e = 0.001$ and $I = 0.05$ deg: $\phi(R)/R^2$ versus R (left), and $\langle v(R) \rangle$ versus R (right). The long-dashed line in the plot at left is the expected value of P_i computed using the methods of Bottke et al. (1994) and Dell’Oro & Paolicchi (1998).

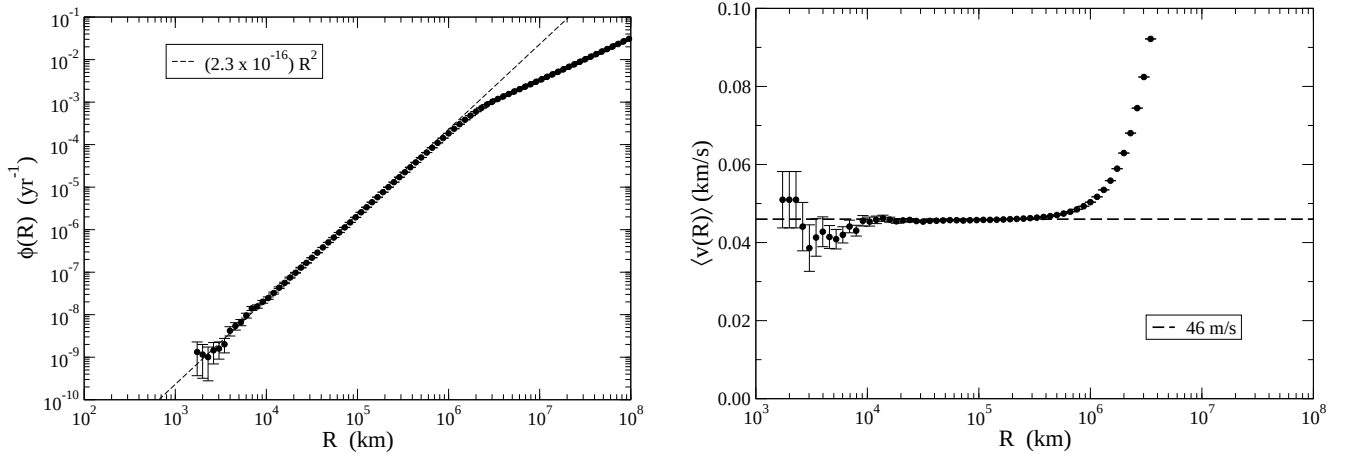


Figure 5. Close encounter rate $\phi(R)$ versus R and mean relative velocity for close encounters within a distance R versus R , computed for a continuous distribution of orbits (see text for details), assuming canonical statistics hypotheses.

as intrinsic probability of collision $P_i = 8.76 \times 10^{-16} \text{ km}^{-2} \text{ yr}^{-1}$ (long-dashed line in Fig. 4 at left), in good agreement with the values of $\phi(R)/R^2$ provided by our algorithm. In particular, the trend $\phi(R) \propto R^2$ is clearly found for R below $\sim 5 \times 10^5$ km. For what concerns the relative velocity, $\langle v(R) \rangle$ rapidly converges to the limit value of 25 m/s, corresponding to the only possible value of the impact velocity. Indeed, the distribution of the impact velocity is a Dirac function, equal to zero except for the value of 25 m/s. This comes from the fact that only two crossing geometries are possible. The two orbits have the same semimajor axis a_0 . Being the first one circular, if they have a point in common it must be at a distance from the Sun equal to a_0 . This entails that the true anomaly f of the crossing point on the eccentric orbit must fulfill the condition $\cos f = -e$. In any point of the circular orbit, the transverse component of the orbital ve-

locity is $v_0 = \sqrt{GM/a_0}$, while the radial component is null. On the eccentric orbit, at the crossing point, the modulus of the orbital velocity is again equal to v_0 , with radial component $v_R = \pm e v_0$ and transverse component $v_T = \sqrt{1 - e^2} v_0$. Being the inclination of the circular orbit null, the crossing point is one of the nodes (ascending or descending) of the eccentric orbit. This means that the direction of the transverse component of the velocity on the eccentric orbit forms an angle I with respect to the transverse component of the circular orbit. In conclusion, the relative velocity U at the crossing point is:

$$\begin{aligned} U^2 &= (v_T \cos I - v_0)^2 + v_R^2 + v_T^2 \sin^2 I = \\ &= 2v_0^2(1 - \cos I \sqrt{1 - e^2}) \end{aligned} \quad (10)$$

From Eq. (10) it results in fact that $U = 25$ m/s.

The last example consists of a continuous distribution of orbits forming a torus with minimum pericenter distance q_{min} and maximum apocenter distance Q_{max} . More precisely, the density distributions of pericenter and apocenter distances (q, Q) is different from zero and uniform in the region of values $q \in [q_{min}, Q_{max}]$ and $Q \in [q_{min}, Q_{max}]$, apart from the values where $q < Q$. The inclinations are uniformly distributed in the interval $[0, I_{max}]$. In this case we no longer deal with orbits with fixed orbital elements, and the sampling model must be changed accordingly. More exactly, it consists of the following step:

- (CT1) the pericenter distance q_A of a particle A is randomly generated between q_{min} and Q_{max} , while its apocenter distance Q_A is randomly generated between q_A and Q_{max} . Moreover, the pericenter distance q_B of a particle B is randomly generated between q_{min} and Q_{max} , while its apocenter distance Q_B is randomly generated between q_B and Q_{max}
- (CT2) the semimajor axes and eccentricities of the particles A and B are obtained from:

$$a_A = \frac{Q_A + q_A}{2} \quad e_A = \frac{Q_A - q_A}{Q_A + q_A} \quad (11)$$

$$a_B = \frac{Q_B + q_B}{2} \quad e_B = \frac{Q_B - q_B}{Q_B + q_B} \quad (12)$$

- (CT3) the inclinations I_A and I_B are randomly generated between 0 and I_{max} ;
- (CT4) the previously defined steps (C1), (C2) and (C3) are executed, where $i = A$ and $j = B$;

Therefore, in this sampling model, every time a new dynamical configuration is generated, *all* orbital elements change, *including* semimajor axes, eccentricities and inclinations. In this case, as explained before, the quantities computed by means of Eq. (1)-(4) are the averages values of ϕ , Ψ and $\langle v \rangle$ taking into account all the possible pairs of orbits (a_A, e_A, I_A) and (a_B, e_B, I_B) generated according to the rules (CT1)-(CT3), assuming uniform circulations of the angles Ω , ω and M of both orbits. Fig. 5 shows $\phi(R)$ and $\langle v(R) \rangle$, versus R , for a torus of orbits with $q_{min} = 2.99$ AU, $Q_{max} = 3.01$ AU and $I_{max} = 0.2$ deg. As comparison, we have computed the mean $\bar{P}_i$ and the mean $\bar{U}_m$ using the methods of Bottke et al. (1994) and Dell'Oro & Paolicchi (1998). We generated a random list of 1,000 orbits (a, e, I) following the steps (CT1)-(CT3). Then for each of the 499,500 possible pairs of orbits we computed $\bar{P}_i$ and $\bar{U}_m$ using the formulas of Bottke et al. (1994). Finally we computed the average values, obtaining $\bar{P}_i = 2.3 \times 10^{-16}$ km⁻² yr⁻¹ and $\bar{U}_m = 46$ m/s. In Fig. 5 at left the long-dashed line represents the function $\bar{P}_i R^2$, in good agreement with the values obtained by our method. The same good agreement results also between $\bar{U}_m$ and the asymptotic value of $\langle v(R) \rangle$ (Fig. 5 at right). At least for $R > 10^4$ km, while below 10^4 km random fluctuations became dominant.

4 SECULAR PERTURBATIONS

In this section we give an example of the use of the

present method in a case for which the dynamical evolution of the orbits does not fulfill the canonical hypotheses of the Öpik/Wetherill statistics. In this case the problem of computing the probability of collision cannot be solved using neither the classical methods based on the approach or the assumption of Öpik/Wetherill nor the more general techniques introduced by Dell'Oro & Paolicchi (1998), for the reason that will be clear soon.

The most important and general case of this type is the statistics of close approaches among objects the orbits of which evolve with time according to the Laplace-Lagrange linear secular theory (Murray & Dermott 1999), that describes the results of the first order perturbations. We consider only the simplest case of two bodies A and B orbiting around a central mass m_0 . The masses m_A and m_B of the two bodies are negligible with respect to m_0 . Moreover the mass m_A is negligible with respect m_B . In this conditions the body B does not feel any gravitational perturbation from A , and its orbit is pure keplerian. On the contrary B is a perturber for A , and the orbit of A is no longer keplerian. The secular theory tell us how the orbit of A changes with time.

We pass over some technical details of the secular theory, and in particular the distinction between mean and osculating elements, that has not much importance for the collision statistics. We use the terms orbital elements or instantaneous orbital element referring to the mean elements. Leaving aside the details of the theory, the features of the evolution of the orbital elements that are relevant for the statistics of collisions are the following. The semimajor axis a of the orbit of A is again a constant of motion and so its value is fixed. The eccentricity varies with time and its value is linked to the value of the longitude of the pericenter. In particular, the evolution of the orbit A is more suitably described by means of the non-singular elements h and k defined as:

$$h = e \sin \tilde{\omega} \quad k = e \cos \tilde{\omega}$$

where e is the eccentricity, and $\tilde{\omega} = \omega + \Omega$ is the longitude of the pericenter. In the plane (k, h) the point P representing the orbit moves with constant angular velocity along a circle of radius e_p (Fig. 6 top left). The quantity e_p is the proper eccentricity. The center of the circle does not coincide in general with the origin of the plane (k, h) . Its distance from the origin is the forced eccentricity e_f , and its position angle with respect the k -axis is the forced pericenter longitude $\tilde{\omega}_f$. The distance of the point P from the origin is the instantaneous eccentricity e while its position angle with respect to the k -axis is the instantaneous pericenter longitude $\tilde{\omega}$. As already anticipated, the position angle α of the point P on the circle changes uniformly with time, i.e. $d\alpha/dt$ is constant. A similar scheme regards the evolution of the inclination I . The corresponding non-singular elements are:

$$P = \sin I \sin \Omega \quad Q = \sin I \cos \Omega$$

In the plane (Q, P) (Fig. 6 top right) the point representing the instantaneous orbit moves with constant angular velocity ($d\beta/dt = \text{const.}$) along a circle of radius $\sin I_p$ (sinus of the proper inclination). The distance between the center of this circle and the origin is the sinus of the force inclina-

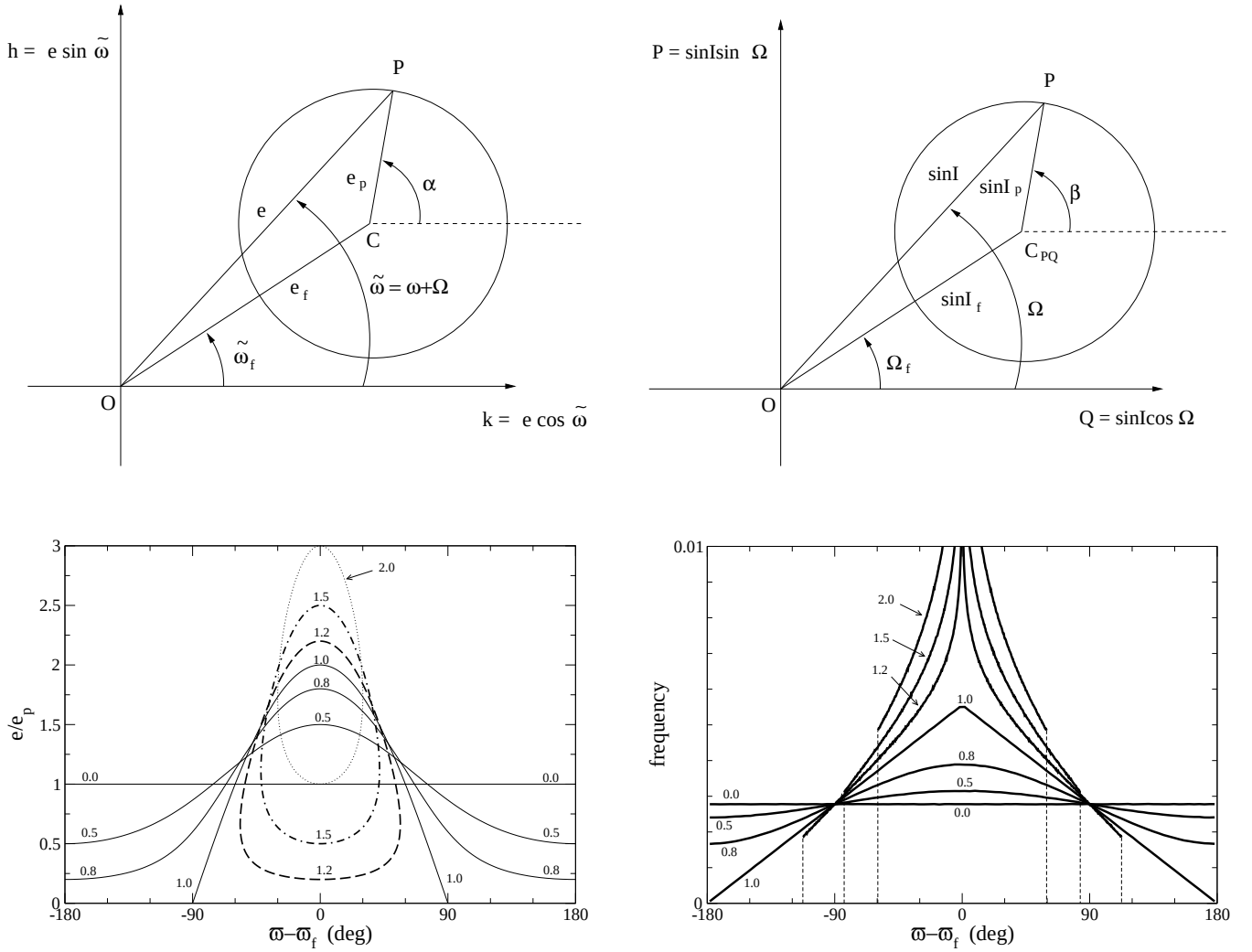


Figure 6. Top left: plane of the non singular elements h and k . Top right: Plane of the non singular elements P and Q . Bottom left: ratio e/e_p versus the difference $\tilde{\omega} - \tilde{\omega}_f$ (the numbers are the values of the ratio e_f/e_p). Bottom right: distribution of the angle $\tilde{\omega} - \tilde{\omega}_f$ (the numbers are the values of the ratio e_f/e_p)

tion I_f , and the position angle of the center is the forced longitude of node Ω_f . For what concerns the instantaneous orbital elements, the distance of the point P from the origin is the sinus of the instantaneous inclination I , and its position angle with respect to the Q -axis is the instantaneous longitude of node Ω .

The forced elements e_f , $\tilde{\omega}_f$, I_f and Ω_f are directly related to the orbital elements of the perturber B , and the semimajor axis of the orbit of A , and so they are fixed. The fact that the representative points P are constrained to move uniformly along the above mentioned circles has the consequence that e and I are no longer constant, and that $d\Omega/dt$ and $d\omega/dt$ are not constant too, violating the fundamental assumptions of the Öpik/Wetherill approach. But more importantly, in this dynamical system correlations between the values of e and $\tilde{\omega}$, and between the values of I and Ω , exist.

This last feature prevents from using the method of Dell'Oro & Paolicchi (1998) for the computation of the collision probabilities between bodies the orbits of which behaves as described above, as well as any other method strictly based on Öpik/Wetherill assumptions, like Kessler

(1981) or Bottke et al. (1994). Although the method of Dell'Oro & Paolicchi (1998) has been developed to treat the case of a generic distribution of the elements ω and Ω , nevertheless it requires that no correlation must exist between them and any of the elements a , e and I . It is worthwhile to note that the importance of the correlation above for what concerns the statistics of collision depends on the values of e_f and I_f with respect to e_p and I_p (Fig. 6, bottom left). In the case the forced elements are small enough compared to the corresponding proper elements, the correlation tends to become weaker and the method of Dell'Oro & Paolicchi (1998) can again be used, as in the case of the large majority of the Main Belt asteroids.

The method described in this paper does not require any assumption about the instantaneous orbital elements of the bodies. So we can use it to study the statistics of collisions among particles the orbit of which are affected by secular perturbations. The system is composed, besides the central mass m_0 , of the perturber B and of a group of test bodies of type A , that are perturbed by B but not perturbing B . In other words we are treating groups of particles the

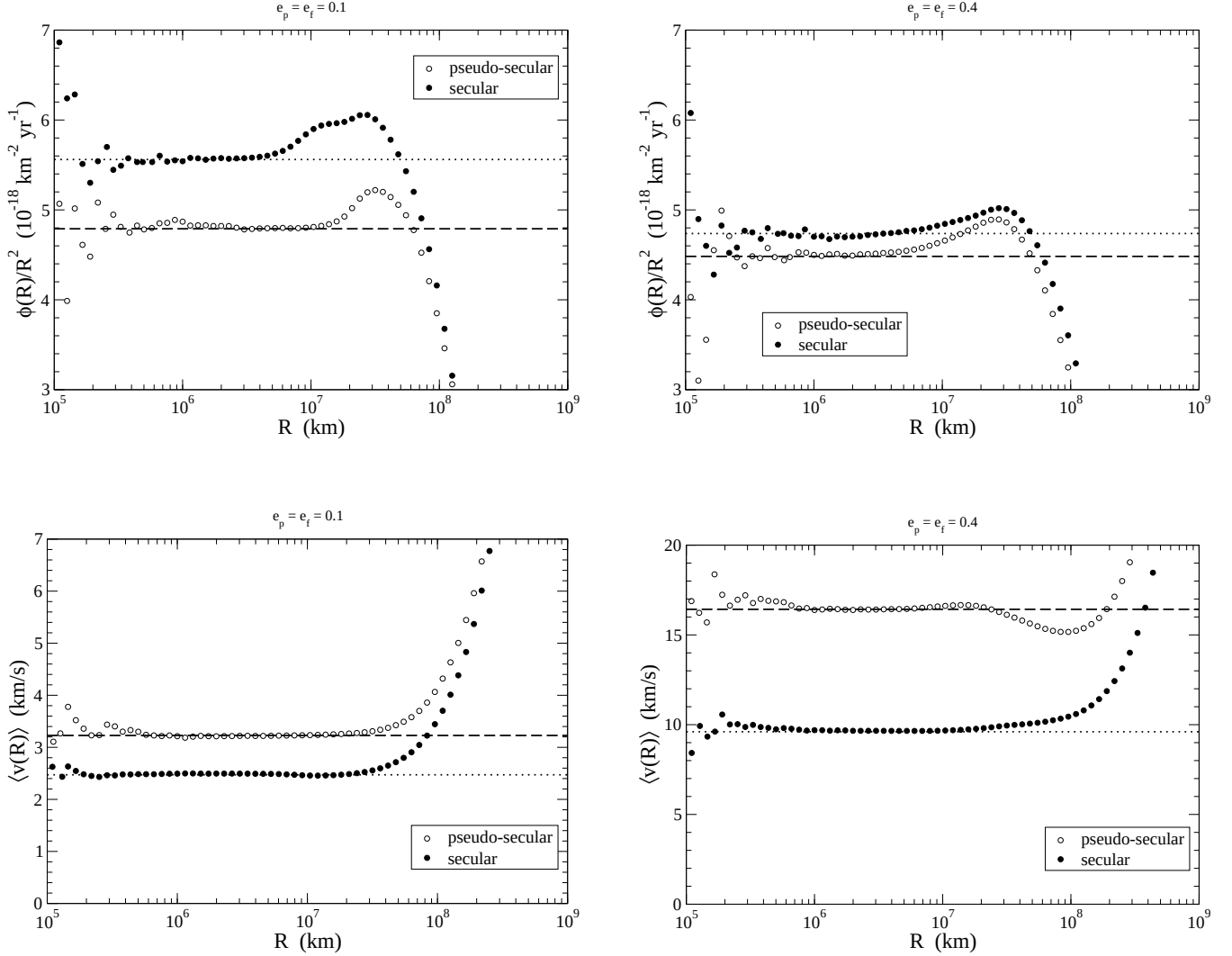


Figure 7. Statistics of collisions between two orbits evolving according to the linear perturbation theory (see text for details). The horizontal long-dashed lines represent the values of P_i and U_m for the pseudo-secular systems computed using the method of Bottke et al. (1994). The dotted lines highlight the asymptotic values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ for $R \rightarrow 0$. On the left the case $e_p = e_f = 0.1$, while on the right the case $e_p = e_f = 0.4$.

orbit of which are affected by the same perturber. We are interested in the statistics of close encounters among the test bodies. Each one of the test bodies is characterized by the fixed values of the parameters a , e_p , e_f , I_p , and I_f while the fixed values of the forced angles $\tilde{\omega}_f$ and Ω_f are the same for all the test bodies, being specific of the orbit of the perturber only. On the other hand, the particular values of $\tilde{\omega}_f$ and Ω_f are not important because, being in common with all the particles, a change of those parameters produces only rigid rotations of the entire system, without affecting the relative geometric configurations of the particle trajectories. The required sampling model for the random generation of the samples (r_k, v_k) of the relative position and the relative velocity between two test bodies, necessary to use Eq. (1)-(3), consists of the following procedure. Once two test particles are chosen, for each of them:

- (S1) random and independent values of the angles α and β are generated in the interval $[0, 2\pi]$;
- (S2) the values of e and $\tilde{\omega}$ are computed on the basis of the fixed parameters e_f , e_p , $\tilde{\omega}_f$ and the current value of α ;
- (S3) the values of I and Ω are computed from the fixed parameters I_f , I_p , Ω_f and the current value of β ;
- (S4) the value of the argument of the pericenter is simply derived as $\omega = \tilde{\omega} - \Omega$;
- (S5) a random value of the mean anomaly M is generated in the interval $[0, 2\pi]$;
- (S6) finally, starting from a , e , I , ω , Ω and M , the vector of the position (x, y, z) and the vector of the velocity $(\dot{x}, \dot{y}, \dot{z})$ are computed;

Such procedure is followed for each particle independently. As done in the previous section, the pair of particles can be fixed once for all or randomly chosen among a list of sets of

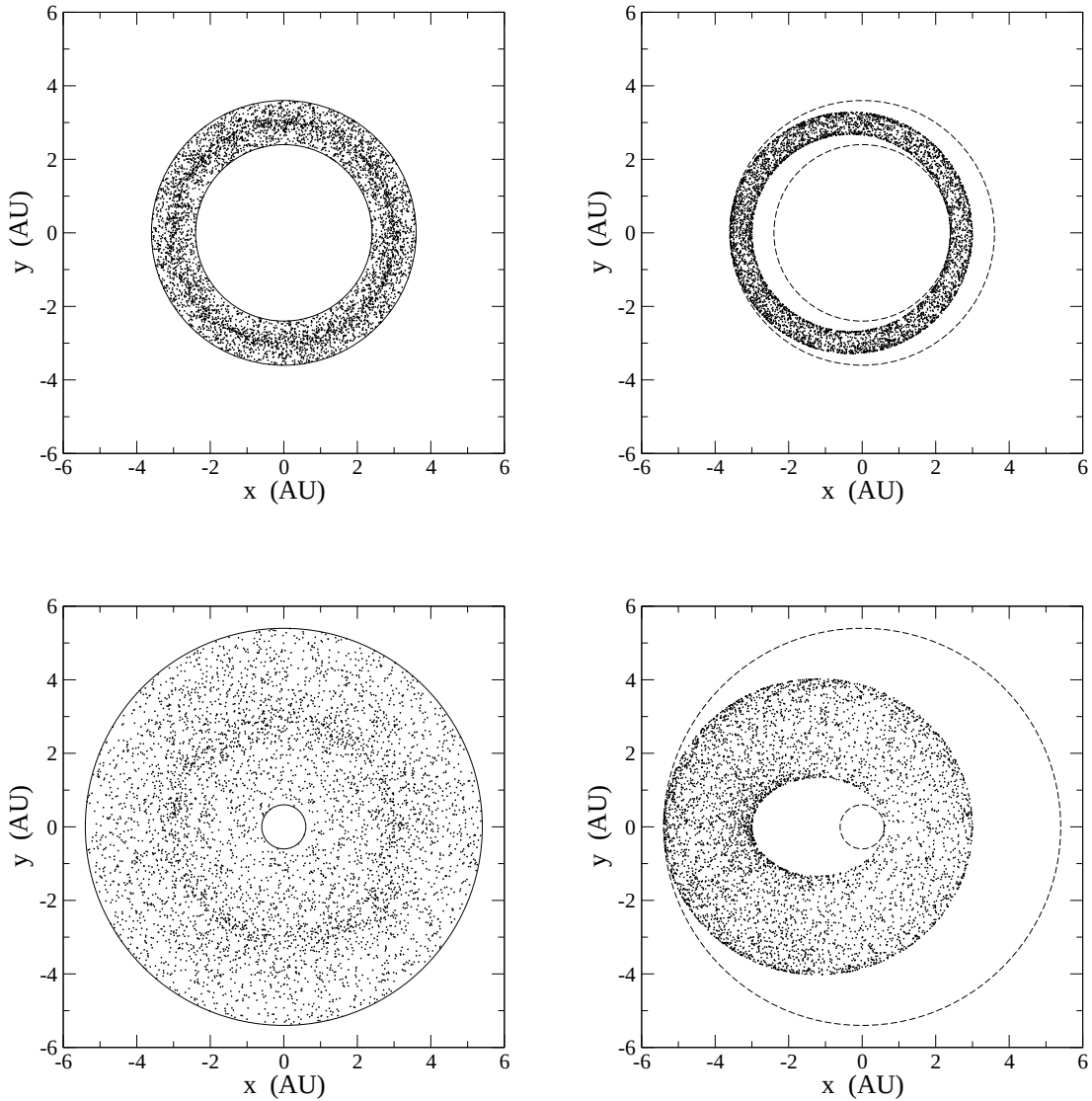


Figure 8. Distribution of the positions of the particles of the pseudo-secular (left) and secular (right) system. Each dot represents the position of the particle projected on the ecliptic plane at a given epoch. At the top the case with $e_p = e_f = 0.1$, at the bottom $e_p = e_f = 0.4$. In the plots at right the long-dashed circles redraw the regions filled by the points in the pseudo-secular system (at left).

elements (a, e_p, e_f, I_p, I_f) . In the second case the final results obtained by Eq. (1)-(3) corresponds to the averages of the quantities $\phi(R)$, $\Psi(R, Q)$ and $\langle q(R) \rangle$ with respect to the distribution of the secular elements of the particles a , e_p , e_f , I_p , and I_f . Once we have obtained the components of the positions and the velocities of the two particles we can determine the relative distance r and the relative velocity v , closing the sample model loop.

We examine an example of such dynamical system consisting in a couple of particles sharing the same perturber. The semimajor axes are set very similar but not identical, and precisely 3.0 AU and 3.05 AU. It is worthwhile to note that if the semimajor axes were exactly the same the sampling model (S1)-(S6) would not reproduce the dynamical properties of the system. In fact, in this case the rate of variation of the mean anomalies would be the same, and the

difference between the two mean anomalies would be constant in time. Consequently, the two mean anomalies cannot be generated independently, as in step (S5). The forced inclinations are set null, for simplicity. Such last choice entails a fixed instantaneous inclination, that in this case are set 2.5 deg and 5 deg. For what concerns the eccentricity, we assume that both particles have the same proper eccentricity and forced eccentricity, and that $e_p = e_f$ for both. We consider the two cases with low proper/forced eccentricity 0.1 and high proper/forced eccentricity 0.4.

Being our main goal here to focus on the effects produced by the correlation between the eccentricity and the longitude of the pericenter, we compare this dynamical system with another pseudo-secular system sharing with the original one all the properties but the $e - \tilde{\omega}$ correlation. The sampling model of this pseudo-secular system is obtained

by that of the secular one previously described inserting just between the steps (S5) and (S6) a “randomization” step (SP5bis) consisting in forgetting the computed values of ω and Ω in steps (S3) and (S4) and substituting them with new values randomly and independently generated in the interval $[0, 2\pi]$. Such modification is equivalent to replace each particle with: (i) an ensemble of pseudo-orbits with the same semimajor axes but with eccentricities and inclinations obtained generating at random the angles α and β , (ii) circulating uniformly, i.e. characterized by a uniform distribution of the angles ω and Ω . The possible values of the eccentricities of these pseudo-orbits belong to the interval $[0, 0.2]$ in the case $e_p = e_f = 0.1$, or $[0, 0.8]$ if $e_p = e_f = 0.4$. In few words, the pseudo-secular system is in reality a perfect canonical system, sharing with the real secular system the same distributions of the osculating eccentricities and inclinations. So, it is correct to use the methods of Bottke et al. (1994) or Dell’Oro & Paolicchi (1998) to compute the probability of collisions among the particles of the pseudo-secular system. Fig. 7 shows the results of this exercise. The values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ versus R for the secular system are represented with black dots, and with white dots for the pseudo-secular one. The horizontal long-dashed lines represent the values of the mean intrinsic probability of collision P_i and the mean impact velocity U_m for the pseudo-secular systems computed using the methods Bottke et al. (1994) and Dell’Oro & Paolicchi (1998). Again, for the pseudo-secular systems, the asymptotic values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ are fully compatible with the corresponding values of P_i and U_m . Instead, the horizontal dotted lines represent the estimation of the asymptotic values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ for $R \rightarrow 0$ in the case of the secular systems (black dots). The scattering of the points for $R \lesssim \times 10^6$ km are numerical random fluctuations of the sampling series of the phase space.

The impact velocities in the pseudo-secular systems are larger than in the corresponding secular systems because in the former case all possible relative orientations among the orbits are allowed, in particular with large and small difference of the angles $\tilde{\omega}$ as well. Instead, in the second case the distribution of the difference of the angles $\tilde{\omega}$ has the triangular shape shown in Fig. 6, bottom right, for the case $e_f/e_p = 1$, with the consequence that the lines of the apses of the orbits tends to be oriented preferably towards the direction $\tilde{\omega} \sim \tilde{\omega}_f$, reducing the relative velocities at the close encounter.

For what concerns the frequency of the close encounters, the intrinsic rate $\phi(R)/R^2$ in the case of secular systems (black dots) results to be systematically larger than the corresponding pseudo-secular case (white dots). This is due again to the non random orientation of the lines of apses, that getting closer the orbits reduces the available volume for the particles. Indeed, as pointed out by Farinella & Davis (1992), the order of magnitude of the intrinsic probability of collisions could be roughly estimated using a simple particle-in-a-box model as $P_i \approx \pi U/W$, where U is the average relative velocity and W the “interaction volume” of the particles. Labeling with the subscript “s” the secular case and with “ps” the pseudo-secular case, we can write $P_s/P_{ps} = (U_s/U_{ps})/(W_s/W_{ps})$. The reduction W_s/W_{ps} of the available volume from the pseudo-secular to the secular case, i.e. from completely random apses orientation to more or less high degree of orbit polarization, is not eas-

ily computable *a priori*. In Fig. 8 we show the distribution of the positions of the two particles, or more exactly of their coordinates (x, y) on the ecliptic plane generated by the sampling model. In the case of pseudo-secular systems the particles move inside a torus-shaped region, centered at the origin of the coordinates (Sun), with internal radius $r_{min} = a(1 - 2e_f)$ and external radius $r_{max} = a(1 + 2e_f)$. In the case of the secular systems, the interaction volume is again a torus not centered around the origin but completely internal to the region filled by the pseudo-secular system. Note that, in the sampling model, we have set $\omega_f = 0$, so the mean line of apses is along the x -axis and the mean pericenter has positive abscissa. The distribution of the coordinate z is the same both in the pseudo-secular and in the corresponding secular system, because we have set $I_f = 0$ in both systems. So the ratio W_s/W_{ps} between the volumes of the secular and pseudo-secular tori are equal to the ratio between the areas of the regions filled with points in Fig. 8 at right and the corresponding areas at left. From the figure it is possible to estimate that $W_s/W_{ps} \simeq 0.5$ in both cases $e_p = e_f = 0.1$ and $e_p = e_f = 0.4$. On the other hand, from Fig. 7 it results than $U_s/U_{ps} \simeq 0.76$ for $e_p = e_f = 0.1$, and $U_s/U_{ps} \simeq 0.58$ for $e_p = e_f = 0.4$. So in conclusion P_s/P_{ps} should be around 1.5 for the case $e_p = e_f = 0.1$, and $P_s/P_{ps} \simeq 1.16$ if $e_p = e_f = 0.4$. Such evaluation, based on a simplistic particle-in-a-box model, overestimates what it results from our simulations, that are respectively ~ 1.16 and ~ 1.05 , but it helps to understand the general trend shown in Fig. 7. In particular that, in this case, the intrinsic probability of collision increases a little taking into account the correlation between the eccentricities and longitude of the pericenter because the reduction of the interaction volume is more than balanced by the reduction of average relative velocity.

We want to stress that the results shown in this Section are not representative of the general features of the statistics of close encounters in systems affected by secular perturbations, but they are only an instructive example of the methodology presented in this paper. We refer to future and more comprehensive works devoted to the effect of the secular perturbations on the statistics of impacts.

5 CONCLUSIONS

In this paper we have presented a general method of study of the impact statistics among asteroids. The method belongs to the family of methods based on a Monte Carlo strategy consisting in deriving the collision rate and the related statistical distributions starting from a random sampling of the phase space. It differs from other Monte Carlo methods already described in literature for the way to derive the rate of close encounters, and the impact probability of collision, from the series of the phase space samplings. The mathematical formalism to do this is very simple and consists only of the computation of sums of sample values of the relative velocity.

The method requires an algorithm of random generation of the relative distance r and relative velocity v of the particles under investigation. This sampling algorithm must correctly reproduce the dynamical behavior of the system, in the sense that repeating a large number of times the random

generation of the couples of values (r, v) , their joint distribution must match the probability distribution to find their values in the intervals $[r, r + dr]$ and $[v, v + dv]$, observing the real system at an instant of time chosen at random. Indeed, the sampling model is not really part of our method, but is rather the prerequisite to use it. The way to sample the phase space depends on the particular case under investigation, and it characterizes the type of collision statistics. For instance, a flat random sampling of the pericenter arguments and node longitudes corresponds to the canonical statistics introduced by Öpik/Wetherill, while the procedure described in Section 4 reproduces the dynamical properties of systems affected by secular perturbations. In both cases, and in any other case we are able to generate correctly the sampling values (r, v) , the Eqs. (1)-(4), that form the basic tools of the method, are always valid.

Eqs. (1)-(4) provide the mean temporal rate $\phi(R)$ of close encounters within a distance R and the distribution Ψ of any parameters related to the kinematic circumstances of the close encounters. The validity of Eqs. (1)-(4) relies on some simplified assumptions that are not valid in general, but that are more and more correct for smaller and smaller values of R , like the linearity of the relative motion of particles during the close encounters. On the other hand, the main goal of this method is not to produce reliable frequencies of close approaches for any value of R . This problem is well beyond the scopes of this work. Instead, the present method is specifically designed for the study of the statistics of impacts among orbiting bodies, the sizes of which are very small compared to the dimensions of their orbits. For this reason, we are interested only in the asymptotic values of $\phi(R)$ and Ψ for $R \rightarrow 0$, for which the underlying assumptions are valid. On the other hand, computing $\phi(R)$ and Ψ for smaller and smaller values of R is more and more numerical prohibitive. So the usual approach is to extrapolate $\phi(R)$ down to values of R comparable with the physical sizes of the bodies under investigation. This extrapolation must be done carefully for two reasons. Two values $\phi(R_1)$ and $\phi(R_2)$ of the mean rate of close encounter computed by means of our method for two different values of R_1 and R_2 of the close encounter distance, with $R_2 > R_1$, are affected by some numerical random errors due to the fluctuations of the phase space sampling. Such errors are not independent because the samplings with $r < R_1$ contribute to the computation of both $\phi(R_1)$ and $\phi(R_2)$ (see Eq. (1)). The correlation between the errors is the larger the more R_1 and R_2 are close. For this reason, in some cases, the random sampling of the phase space produce false trends in the function $\phi(R)$ within intervals of under-sampled values of R , corresponding to large errors of $\phi(R)$, like that shown in Fig. 2. The second reason of caution in extrapolating comes from the fact that, in general, there is no guarantee that the trend $\phi(R) \propto R^2$ really holds for $R \rightarrow 0$. This can happen for two nearly tangent but not crossing orbits, for which $\phi(R) = 0$ for values of R smaller than the minimum possible distance between the orbits, and that can be hardly explored by means of a random sampling. For these reasons, the extrapolation should be joint to a careful check of the possibility of intersection of the trajectories, identifying the case for which an orbital crossing is not possible for geometrical reasons, or because of the effect of dynamical protection mechanism (like that

of the Neptune-Pluto system). This topic will be studied in details in a forthcoming paper.

Concerning the hypothesis about the asymptotic trend of ϕ for $R \rightarrow 0$, it is worthwhile to note that some cases exist for which $\lim_{R \rightarrow 0} \phi(R) \propto R^\gamma$ with $\gamma \neq 2$ (Milani et al. 1990; Vedder 1996). An example consists of two coplanar orbits with zero inclination, where for small values of R the rate ϕ is proportional to R instead of R^2 , simply because the third dimension is missing and we have a perfectly bi-dimensional problem. In this case the relevant parameter is no longer $\phi(R)/R^2$ but rather the ratio $\phi(R)/R$, the limit of which for $R \rightarrow 0$ is a kind of intrinsic probability of collision measured in $\text{yr}^{-1} \text{ km}^{-1}$. Even if in this paper we focused on the “standard” case $\gamma = 2$, the method can be easily adapted to other situations, as briefly mentioned in Appendix A.

Like any Monte Carlo approach, even this method has the major drawback to be computationally slow. In general, the method requires a very large number of samples. However, in the case of asteroids, it is prohibitive to obtain a reliable estimation of $\phi(R)$ for values of the order of 1-100 km. In the examples shown in this work, we explored the function $\phi(R)$ for values of R larger than few 10^5 km, but however enough to obtain the value of the intrinsic probability of collision, the mean impact velocity and its frequency distribution with acceptable accuracy. The computational effort depends on the number of samples necessary to sample densely enough the phase space, that in turn depends on the dynamical and geometrical properties of the system under investigation. The computational time depends on the kind of hardware at disposal and the way the sampling model simulating the system is implemented. For instance, our software implementation of the sampling model for the case of canonical statistics (Section 3) is able to generate 10^9 samplings in about 0.56 hours using one single Intel i3-2100 CPU. The values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ reported in Table F3 of Appendix F2 have been obtained with simulations of 10^9 samplings for each of the targets in the list. The distributions shown in Fig. 3 at the top and at bottom left have been obtained with a simulation of 10^{10} samplings (about five hours), while the distribution at the bottom left required a much longer computational time (a twenty-day long simulation consisting of 10^{12} samplings). On the other hand, as we have seen, the computational effort (and time) does not depend on the number of particles forming the system under investigation if we are interested on the mean value of the intrinsic probability of collision and the average impact velocity distribution. The plots in Fig. 5 showing $\phi(R)$ and $\langle v(R) \rangle$ versus R for the case of a continuous distribution of orbits required the same computational time as for the single pair of orbits of Fig. 1. For this reason the Monte Carlo approach is particularly suitable for the study of the statistics of impacts in systems consisting of large numbers of particles, or even described by continuous distributions of mass.

The major advantage of a Monte Carlo approach consists in flexibility together with robustness and ease of implementation. Methods based on the analytical description of the geometry of orbital crossing require the developments of formulas (mainly integrals) specifically valid for a particular type of evolution of the orbits. Also in the general method of Dell’Oro & Paolicchi (1998) the analytical form of the density distribution of the angular orbital elements has to be

provided explicitly. This in some cases can be very difficult. Instead, in many cases is much more easy to develop an algorithm of numerical generation of the same orbital elements. Indeed, the general formulas of Section 2 are valid regardless the way the samples (r, v) are generated, provided that such samplings reproduce faithfully the dynamics of the system.

We have discussed the formulation of the sampling model in the simple but important case of orbits with evolution governed by the canonical Öpik/Wetherill hypotheses (problem already solved analytically), and then we have extended the sampling model to include the secular perturbations produced by one perturbing body. Other kinds of dynamical systems could be treated if the corresponding sampling models are correctly developed. The extension of the sampling model described in Section 4 to the case of secular perturbations due to two or more perturbing bodies is straightforward, because it simply consists of the superposition of further epicycles in the planes of the variables (h, k) and (P, Q) . The dynamical behavior of asteroids locked in some kinds of resonance, like mean motion, secular or Kozai resonances, can be reproduced if the orbital elements are randomly generated but taking into account the constraints due to the critical arguments characterizing the specific resonances. In this case the constraints manifest as statistical correlations among the elements involved in the definition of the critical arguments. However we do not intend discussing those cases here. The main aim of this paper is only to provide the basic principles of the method, devoting forthcoming papers to the study and detailed discussion of the statistics of impacts in specific dynamical systems.

REFERENCES

- Bottke W.F., Greenberg R., 1993, *Geophys. Res. Lett.*, 20, 879
 Bottke W.F., Nolan M.C., Greenberg R., Kolvoord R.A., 1994, *Icarus*, 107, 255
 Dahlgren M., 1998, *A&A*, 336, 1056
 Dell’Oro A., Campo Bagatin A., Benavidez P.G., Alemañ R.A., 2013, *A&A*, 558, A95
 Dell’Oro A., Marzari F., Paolicchi P., Vanzani V., 2001, *A&A*, 366, 1053
 Dell’Oro A., Paolicchi P., 1997, *Planet. Space Sci.*, 45, 779
 Dell’Oro A., Paolicchi P., 1998, *Icarus*, 136, 328
 Dell’Oro A., Paolicchi P., Cellino A., Zappalà V., 2002, *Icarus*, 156, 191
 Dell’Oro A., Paolicchi P., Marzari F., Dotto E., Vanzani V., 1998, *A&A*, 339, 272
 Farinella P., Davis D.R., 1992, *Icarus*, 97, 111
 Greenberg R., 1982, *AJ*, 87, 185
 Horner J., Jones B.W., 2008, *International Journal of Astrobiology*, 7, 251
 JeongAhn Y., Malhotra R., 2015, *Icarus*, 262, 140
 Kessler D.J. 1981, *Icarus*, 48, 39
 Marzari F., Scholl H., Farinella P., 1996, *Icarus*, 119, 192
 Milani A., Carpino M., Marzari F. 1990, *Icarus*, 88, 292
 Milani A., Knežević Z., 1990, *Celestial Mechanics and Dynamical Astronomy*, 49, 247
 Murray C.D., Dermott S.F., 1999, *Solar system dynamics*
 Öpik E.J., 1951, *Proc. of the Royal Irish Academy Section A*, 54, 165
 Pokorný P., Vokrouhlický D., 2013, *Icarus*, 226, 682
 Rickman H., Wiśniowski T., Wajer P., Gabryszewski R., Valsecchi G.B., 2014, *A&A*, 569, A47
 Vedder J.D., 1996, *Icarus*, 123, 436

Vedder J.D., 1998, *Icarus*, 131, 283

Vokrouhlický D., Pokorný P., Nesvorný D., 2012, *Icarus*, 219, 150

Wetherill G.W., 1967, *J. Geophys. Res.*, 72, 2429

Wiśniowski T., Rickman H., 2013, *Acta Astronomica*, 63, 293

APPENDIX A: MEAN TRANSIT TIME ACROSS A SPHERE

One of basic assumptions of the approach followed in this paper is that during the interval of time in which the relative distance between the two particles is smaller than R , the relative motion is rectilinear with constant velocity. In a reference frame where one particle (target) is at rest and placed at the origin of a cartesian system of coordinates x, y and z , the position $\mathbf{r} = (x, y, z)$ of the other particle (projectile) at the time t is $\mathbf{r} = \mathbf{v}t + \mathbf{r}_0$, where $\mathbf{r}_0$ is the position at $t = 0$. The time t_m corresponding to the minimum value of $\mathbf{r}$ is $t_m = -(\mathbf{v} \cdot \mathbf{r}_0)/v^2$, where v is the modulus of $\mathbf{v}$. For $t = t_m$ the position of the projectile is $\mathbf{b} = \mathbf{r}_0 - (\hat{\mathbf{n}} \cdot \mathbf{r}_0)\hat{\mathbf{n}}$, where $\hat{\mathbf{n}}$ is the unit vector such that $\mathbf{v} = v\hat{\mathbf{n}}$, that is $\hat{\mathbf{n}}$ is the direction of the velocity. It is simple to see that $\mathbf{b}$ is always perpendicular to $\hat{\mathbf{n}}$, and it expresses the impact parameter of the close encounter. In other words, $\mathbf{b}$ is a vector belonging to the plane $\mathcal{P}$ passing through the origin and perpendicular to $\hat{\mathbf{n}}$. From now on we choose as $t = 0$ the epoch t_m , and the motion of the projectile is given as $\mathbf{r} = v\hat{\mathbf{n}}t + \mathbf{b}$.

Let us now imagine that we know only the value v , but not the other details about the motion of the projectile, that is neither the direction $\hat{\mathbf{n}}$ of the velocity vector nor the initial position $\mathbf{b}$. So we assume that (1) the direction of the velocity is isotropically random and (2) the points described by the vector $\mathbf{b}$ are randomly but uniformly distributed on the plane $\mathcal{P}$. The hypothesis (2) is simply the expectation that the transit probability is proportional to the cross-section area. In the hypothesis that the modulus of $\mathbf{b}$ is equal or smaller than R , i.e. that the trajectory of the projectile really intersects the sphere of radius R centered on the origin, we want to compute the average crossing time. Being fixed the module of the velocity, this problem is equivalent to compute the average length of the part of the trajectory inside the sphere, i.e. the length of the chord.

The length s of the chord depends only on the modulus of $\mathbf{b}$ regardless the direction $\hat{\mathbf{n}}$, thanks to the spherical symmetry of the system. So, putting $\hat{\mathbf{n}} = (0, 0, 1)$, the plane $\mathcal{P}$ is the x - y plane and $\mathbf{b} = (x, y, 0)$, where x and y are any values such that $x^2 + y^2 < R^2$ (Fig. A1). The length of the chord is $s = 2(R^2 - b^2)^{1/2}$, where $b^2 = x^2 + y^2$. The coordinates x and y are independent random deviates with uniform density distribution per unit area $\delta(x, y) = 1/(\pi R^2)$. Therefore, the mean value of s is expressed by the integral:

$$\bar{s} = \frac{2}{\pi R^2} \int \int_{x^2 + y^2 < R^2} (R^2 - b^2)^{1/2} dx dy \quad (\text{A1})$$

where the region of integration is the circle of radius R around the origin. This integral can be easily solved in polar coordinates θ and b , where $x = b \cos \theta$ and $y = b \sin \theta$:

$$\bar{s} = \frac{2}{\pi R^2} \int_0^R \int_0^{2\pi} b(R^2 - b^2)^{1/2} d\theta dr =$$

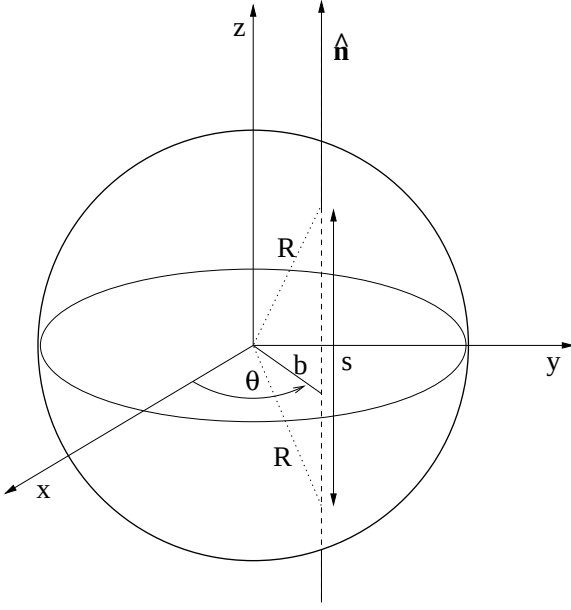


Figure A1.

$$= \frac{4}{\pi R^2} \int_0^R b(R^2 - b^2)^{1/2} db = \frac{4}{3} R \quad (\text{A2})$$

where it has been taken into account that the Jacobian of the variables transformation is $\partial(x, y)/\partial(\theta, b) = b$.

In conclusion the mean duration of the transit inside the sphere of radius R is:

$$\overline{\Delta t} = \frac{4}{3} \frac{R}{v} \quad (\text{A3})$$

It is worthwhile to mention the special case of two-dimension systems, where the sphere is substituted with a circle of radius R around the origin in the plane x - y . In a similar way as before, we can consider the family of lines parallel to the y axis, naming x the abscissa of the point of their intersection with the x axis. In this case the length s of the chord inside the circle is $s = 2(R^2 - x^2)^{1/2}$. Assuming that the quantity x is a random variate with uniform density distribution $\delta(x) = 1/(2R)$ in the interval $[-R, +R]$, the mean value of s is:

$$\bar{s} = \int_{-R}^{+R} \delta(x) dx = \frac{1}{R} \int_{-R}^{+R} (R^2 - x^2)^{1/2} dx = \frac{\pi}{2} R \quad (\text{A4})$$

entailing a mean transit time inside the circle $\overline{\Delta t} = (\pi/2)(R/v)$.

APPENDIX B: RATE OF CLOSE ENCOUNTERS

Let us consider two particles, both moving inside a limited region of space. This means that the particles cannot get away to infinity. Let $r(t)$ be the relative distance between the particles at the epoch t . We are interested in the close encounters within a distance R , i.e. in events during which the two particles pass close to each other within a distance

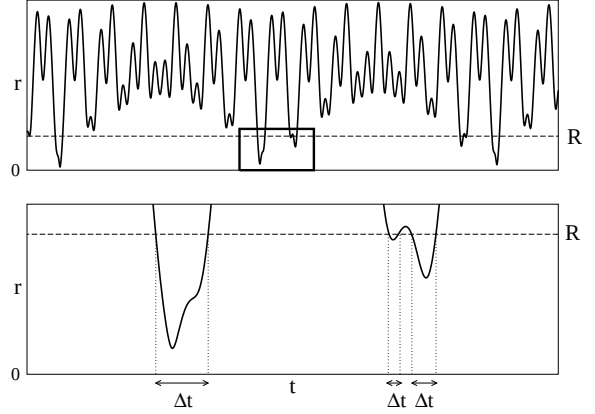


Figure B1. Example of function $r(t)$ for two points moving along keplerian orbits. At the bottom some intervals of time during which $r < R$ are highlighted.

R . In fact, being r a function of time, a close encounter occurs at the epoch t' if $r(t) < R$ for $t \in [t', t' + \Delta t]$, and $r(t) > R$ for t just before t' and just after $t' + \Delta t$ (Fig. B1). The quantity Δt is the duration of the close encounter. It is worthwhile to note that the ensemble of the close encounters within a distance R includes also the ensemble of close encounter within a distance $R' < R$.

During an interval of time T a given number $N(R)$ of close encounters within a mutual distance $r < R$ occur. By definition, the mean temporal rate of the close encounters is:

$$\phi(R) = \frac{N(R)}{T} \quad (\text{B1})$$

We can rewrite this quantity in the following way:

$$\phi(R) = \frac{N(R)}{S(R)} \frac{S(R)}{T} = \frac{p(R)}{\tau(R)} \quad (\text{B2})$$

where $S(R)$ is the sum of the durations of the close encounters that occurred within a distance R during the interval of time T . Or in other words $S(R)$ is the total amount of time during which the mutual distance between the two particles is less than R . Therefore, the ratio:

$$p(R) = \frac{S(R)}{T} \quad (\text{B3})$$

is the probability to find, at any randomly chosen instant of time, the two particles at a distance smaller than R . As for the ratio:

$$\tau(R) = \frac{S(R)}{N(R)} \quad (\text{B4})$$

it is the average duration of close encounters within a distance R .

Moreover we assume that during the interval of time when the mutual distance r is less than R , the relative motion is rectilinear with constant velocity v . However the relative velocity v has a different value in each close encounter. Let $dN(R, v)$ be the number of close encounters within a

distance R and with relative velocity within the infinitesimal interval $[v, v + dv]$. So we can define the mean temporal rate of close encounters with velocity v as:

$$d\phi(R, v) = \frac{dN(R, v)}{T} \quad (\text{B5})$$

This quantity can be rewritten as:

$$d\phi(R, v) = \frac{dN(R, v)}{dS(R, v)} \frac{dS(R, v)}{T} \quad (\text{B6})$$

where $dS(R, v)$ is the sum of the durations of the close encounters with relative velocity v and $r < R$. As before, the ratio:

$$\tau(R, v) = \frac{dS(R, v)}{dN(R, v)} \quad (\text{B7})$$

is the mean value of the duration of the close encounters with velocity v and within a distance $r < R$. From Eq. (A3) we can put:

$$\tau(R, v) = \frac{4}{3} \frac{R}{v} \quad (\text{B8})$$

Moreover:

$$dp(R, v) = \frac{dS(R, v)}{T} \quad (\text{B9})$$

is the probability to find, *at any randomly chosen instant of time*, the two particles at a distance r smaller than R and with relative velocity in the infinitesimal interval $[v, v + dv]$.

Taking into account that:

$$\phi(R) = \int_0^\infty \frac{d\phi(R, v)}{dv} dv$$

from Eqs. (B6), (B8) and (B9) it follows that:

$$\phi(R) = \frac{3}{4} \frac{1}{R} \int_0^\infty v \frac{dp(R, v)}{dv} dv \quad (\text{B10})$$

It is worthwhile to note that $\phi(R)$ is not proportional to $p(R)$. While $\phi(R)$ is the *temporal* rate of close encounters, or in other words $\phi(R)$ is the mean number of times per unit of time that $r(t)$ becomes less than R , $p(R)$ is simply the probability that $r(t) < R$ if the epoch t is chosen at random. Let us imagine a simplified situation in which N close encounters occur during the interval of time T , and the duration Δt is the same for all events. In this case $\phi = N/T$ and $p = N\Delta t/T$. If the duration of close encounters was shorter (for instance if the particles were faster) the probability p would be smaller, but the frequency ϕ would be the same.

In conclusion from (B10) we can compute correctly the temporal frequency of the close encounters within a distance R if we are able to evaluate the probability $dp(R, v)$ to find the two particles at a mutual distance smaller than R and with a relative velocity in the interval $[v, v + dv]$. This goal can be easily reached if we have at our disposal a suitable sampling model correctly describing the dynamics of the system during the interval of time T . A sampling model is an

algorithm able to generate the mutual distance r and the relative velocity v at a instant of time chosen at random. Provided that, let us imagine to generate a series of n independent pairs (r_k, v_k) using our sampling model. Let $dn(R, v)$ be the number of cases for which $r_k < R$ and $v_k \in [v, v + dv]$. Then, by definition of the probability as the ratio between the number of favorable outcomes and the total number of possible outcomes:

$$\frac{dn(R, v)}{n} = dp(R, v) \quad (\text{B11})$$

Substituting this expression of $dp(R, v)$ in Eq. (B10) we can write:

$$\phi(R) = \frac{3}{4} \frac{1}{R} \frac{1}{n} \int_0^\infty v \frac{dn(R, v)}{dv} dv \quad (\text{B12})$$

but, being $v dn(R, v)$ nothing else than the sum of the sample values v_k for which $r_k < R$ and $v_k \in [v, v + dv]$, it is easy to realize that the integral is simply the sum of all generated values v_k for which $r_k < R$. Therefore:

$$\phi(R) = \frac{3}{4} \frac{1}{R} \frac{1}{n} \sum_{k:r_k < R} v_k = \frac{3}{4} \frac{1}{R} \frac{w(R)}{n} \quad (\text{B13})$$

where:

$$w(R) = \sum_{k:r_k < R} v_k \quad (\text{B14})$$

In conclusion the mean temporal rate of close approaches at a distance r smaller than R is simply related to the ratio between the sum of sample values v_k for which $r_k < R$ and the total number of samples. The ratio $w(R)/n$ is not the average of the values v_k because the number of terms of the sum $w(R)$ is lower than n . Alternatively, it is possible to re-write the last formula in this way:

$$\phi(R) = \frac{3}{4} \frac{1}{R} \left(\frac{w(R)}{n(R)} \right) \left(\frac{n(R)}{n} \right) \quad (\text{B15})$$

where $n(R)$ is the number of samples for which $r_k < R$, and $w(R)/n(R)$ is the mean value of the sample velocities v_k .

Finally we note that in the case of a bi-dimensional system the factor $3/(4R)$ in Eq. (B10) and Eq. (B13) is replaced by $2/(\pi R)$.

APPENDIX C: CLOSE ENCOUNTERS DISTRIBUTIONS

Usually we are not interested only in the temporal rate of close encounters, but we need to know the statistical distribution of the values of any parameter q related to the close encounters. As for the relative velocity, we assume that the value of the parameter q does not change during each close approach. So, among all close encounters occurring during the interval of time T , we select those with $r < R$ and q belonging to an ensemble of values $\mathcal{Q}$, and we call $N(R, \mathcal{Q})$ their number. The corresponding temporal frequency is:

$$\phi(R, \mathcal{Q}) = \frac{N(R, \mathcal{Q})}{T} \quad (\text{C1})$$

For the subset of the close encounters with $q \in \mathcal{Q}$ the Eq. (B10) can be generalized as:

$$\phi(R, \mathcal{Q}) = \frac{3}{4} \frac{1}{R} \int_0^\infty v \frac{dp(R, v, \mathcal{Q})}{dv} dv \quad (\text{C2})$$

where $dp(R, v, \mathcal{Q})$ is the probability to find, at any instant of time, the system with $r < R$, $v \in [v, v + dv]$ and $q \in \mathcal{Q}$.

The fraction of events per unit of time with $r < R$ and $q \in \mathcal{Q}$, is:

$$\Psi(R, \mathcal{Q}) = \frac{\phi(R, \mathcal{Q})}{\phi(R)} \quad (\text{C3})$$

or in other words, $\Psi(R, \mathcal{Q})$ is the statistical distribution of the variable q . The corresponding continuous distribution is obtained putting the ensembles $\mathcal{Q}$ infinitesimal.

In order to evaluate the expected value of the parameter, let us consider the total ensemble of all values of q . If $d\Psi(R, q) = d\phi(R, q)/\phi(R)$ is the frequency of the close approaches with the value of the parameter between q and $q + dq$, the expected value of q is by definition:

$$\langle q(R) \rangle = \int q \frac{d\Psi(R, q)}{dq} dq \quad (\text{C4})$$

where the symbol $\langle q(R) \rangle$ highlights that we are talking about the mean value of the parameter q for the close encounter within a distance R . So from Eqs. (C3) and (C2) we have:

$$\langle q(R) \rangle = \frac{\int \int q v \frac{dp(R, v, q)}{dq dv} dq dv}{\int v \frac{dp(R, v)}{dv} dv} \quad (\text{C5})$$

It is worthwhile to note that the true statistical distribution $\Psi(R, \mathcal{Q})$ is proportional to $\phi(R, \mathcal{Q})$ and not to the probability $p(R, \mathcal{Q})$. The quantity $p(R, \mathcal{Q})$ is not the temporal frequency or better the mean number of times per unit of time that $r(t)$ becomes smaller than R while $q \in \mathcal{Q}$; $p(R, \mathcal{Q})$ is only the probability that, choosing at random the epoch t , $r < R$ and $q \in \mathcal{Q}$. Indeed, for a given ϕ , i.e. for a given number of favorable events during the period T , p increases if the corresponding durations Δt increase (for example if the relative velocity during the close encounters is smaller), but ϕ does not change. For instance, in a simplified case, let us imagine to have only two different kinds of close approaches during the interval T : $N/2$ events of type A for which $q = q_A$ and duration $\Delta t = 1$, and $N/2$ events of type B with duration $\Delta t = 2$ and $q = q_B$. If we choose at random an instant of time, the probability to fall in an event A is half of the probability to fall in an event B. However the temporal frequency of the two kinds of events is the same, that is $N/(2T)$. And, the expected value of q is $(q_A + q_B)/2$ and not $(q_A + 2q_B)/3$.

We can evaluate numerically the integrals (C2) and (C5) using a sampling model, following the same approach described in Appendix B to evaluate the temporal frequency $\phi(R)$. This time our sampling model has to provide for each sample pair (r_k, v_k) the corresponding sample value q_k of the parameter. If, using the sample model, we generate a random sequence of n triplets (r_k, v_k, q_k) , with n large enough,

the probability to find $r_k < R$, $v_k \in [v, v + dv]$ and $q_k \in \mathcal{Q}$ is given by:

$$dp(R, v, \mathcal{Q}) = \frac{dn(R, v, \mathcal{Q})}{n} \quad (\text{C6})$$

where $dn(R, v, \mathcal{Q})$ is the number of the random samples with $r_k < R$, $v_k \in [v, v + dv]$ and $q_k \in \mathcal{Q}$. So, substituting in (C2):

$$\phi(R, \mathcal{Q}) = \frac{3}{4} \frac{1}{R} \int_0^\infty v \frac{dn(R, v, \mathcal{Q})}{dv} dv \quad (\text{C7})$$

but again the integral is simply the sum of sample values of the velocity v_k for which $r_k < R$ and $q_k \in \mathcal{Q}$. Therefore:

$$\phi(R, \mathcal{Q}) = \frac{3}{4} \frac{1}{R} \frac{w(R, \mathcal{Q})}{n} \quad (\text{C8})$$

where:

$$w(R, \mathcal{Q}) = \sum_{k: r_k < R, q_k \in \mathcal{Q}} v_k \quad (\text{C9})$$

In conclusion, substituting Eq. (C8) and Eq. (B13) into Eq. (C3), we have:

$$\Psi(R, \mathcal{Q}) = \frac{\sum_{k: r_k < R, q_k \in \mathcal{Q}} v_k}{\sum_{k: r_k < R} v_k} \quad (\text{C10})$$

For instance, the frequency of close approaches with relative velocity belonging to an interval of values Δv is obtained as:

$$\phi(R, \Delta v) = \frac{3}{4} \frac{1}{R} \frac{\sum_{k: r_k < R, v_k \in \Delta v} v_k}{n} \quad (\text{C11})$$

and consequently the normalized frequency distribution of the relative velocity built with respect to a series of bins Δv_j is $\Psi(R, \Delta v_j) = \phi(R, \Delta v_j)/\phi(R)$.

Finally, for what concerns the expected value of q , from Eq. (C5) it is easy to see that:

$$\langle q(R) \rangle = \frac{\sum_{k: r_k < R} q_k v_k}{\sum_{k: r_k < R} v_k} \quad (\text{C12})$$

where the final sums include all the generated pairs (r_k, v_k) with $r_k < R$, and q_k is the corresponding value of the parameter for each pair. Summarizing, the expected value of a parameter q for the close approaches within a distance R is simply the weighted average of the sample values q_k by the corresponding relative velocity.

Important cases of expected values are the mean value of the relative velocity and its square:

$$\langle v(R) \rangle = \frac{\sum_{k: r_k < R} v_k^2}{\sum_{k: r_k < R} v_k} \quad (\text{C13})$$

$$\langle v^2(R) \rangle = \frac{\sum_{k:r_k < R} v_k^3}{\sum_{k:r_k < R} v_k} \quad (\text{C14})$$

APPENDIX D: MULTIPLE PAIRS OF PARTICLES

In Appendix B and C we focused to the case of close encounters between two particles only. If we are interested in the statistics of close approaches among m different particles, the important quantity to be determined is the mean rate of close approaches, defined as the average of the rates for each possible couple of particles. It is given by:

$$\overline{\phi(R)} = \frac{1}{m(m-1)} \sum_{i \neq j} \phi_{ij}(R) \quad (\text{D1})$$

where $\phi_{ij}(R)$ is the rate of close encounters between the i -th particle and the j -th particle. It is important to note that, by definition, also the pairs for which no close encounter within a distance R is possible contribute to the mean rate of close encounters with null terms in the sum (D1).

At first sight, it seems to be necessary evaluating one by one the rates $\phi_{ij}(R)$ for each couple (i, j) of particles before to compute the overall average $\overline{\phi(R)}$. This is not necessarily true.

Let us consider a list of n_{ij} sample pairs of relative distances and velocities (r_{ijk}, v_{ijk}) generated by a suitable sampling model for the couple of orbits of i -th and the j -th particles. From Eq. (B13) we know that:

$$\phi_{ij}(R) = \frac{3}{4} \frac{1}{R} \frac{w_{ij}(R)}{n_{ij}} \quad (\text{D2})$$

where:

$$w_{ij} = \sum_{k:r_{ijk} < R} v_{ijk} \quad (\text{D3})$$

Substituting into (D1) we obtain:

$$\overline{\phi(R)} = \frac{3}{4} \frac{1}{R} \frac{1}{m(m-1)} \sum_{i \neq j} \frac{1}{n_{ij}} \sum_{k:r_{ijk} < R} v_{ijk} \quad (\text{D4})$$

If, for each pair of particles (i, j) , the corresponding list of random samplings (r_{ijk}, v_{ijk}) contains the same number of samples $n_{ij} = \bar{n}$, the last equation becomes:

$$\overline{\phi(R)} = \frac{3}{4} \frac{1}{R} \frac{1}{n} \sum_{k:r_{ijk} < R} v_{ijk} \quad (\text{D5})$$

where n is:

$$n = m(m-1)\bar{n} \quad (\text{D6})$$

the total number of samples for all the particle pairs.

Equation (D5) is formally similar to (B13), but it is valid only if the random samples are evenly allocated among

all the $m(m-1)$ possible pairs of particles. From an operative point of view, we can generate all the samples (r_{ijk}, v_{ijk}) for all the couples of particles (i, j) in a unique sequence, but provided that every time we generate a different sample we chose randomly the two particles i and j (with $i \neq j$). In this way each pair (i, j) has the same probability to be selected, and the number of samples n_{ij} devoted to each pair is in mean the same for all the pairs.

Another important case regards the statistics of close encounters between a given test particle (target) and an ensemble of other particles (impactors). If the number of impactors is m , the mean rate of close encounter is:

$$\overline{\phi(R)} = \frac{1}{m} \sum_i \phi_i(R) \quad (\text{D7})$$

where $\phi_i(R)$ is the rate of close encounters between the target and the i -th impactor. In this case, the analogous formula to (D5) is:

$$\overline{\phi(R)} = \frac{3}{4} \frac{1}{R} \frac{1}{n} \sum_{k:r_{ik} < R} v_{ik} \quad (\text{D8})$$

provided that any time we generate the sample (r_{ik}, v_{ik}) the index i of the of the impactor is randomly chosen, and n is again the total number of samples.

As discussed in detail in the Appendix E, the value of $\phi(R)$ provided by Eq. B13 is not deterministic but it is rather a random deviate fluctuating around a mean value. Both Eq. (D1) and (D7) are special cases of the general expression:

$$\overline{\phi(R)} = \frac{1}{M} \sum_{i=1}^M \phi_i(R) \quad (\text{D9})$$

where M is the number of terms of the sum, that in the two previous examples can be equal to $m(m-1)$ or m . So $\overline{\phi(R)}$ is a random deviate too. From Eq. (E12), the variance of $\phi(R)$ is inversely proportional to the number of samples used to evaluate it. Therefore the variance σ_i^2 of each term $\phi_i(R)$ can be written as c_i^2/n_i , where c_i^2 is a constant and n_i the number of samples used to compute $\phi_i(R)$. Being the fluctuations of one term not correlated with the fluctuations of another one, the variance of $\overline{\phi(R)}$ is:

$$\sigma^2 = \frac{1}{M^2} \sum_{i=1}^M \sigma_i^2 = \frac{1}{M^2} \sum_{i=1}^M \frac{c_i^2}{n_i} \quad (\text{D10})$$

If for all terms $n_i = \bar{n}$, the last equation can be rewritten as:

$$\sigma^2 = \frac{1}{n} \left(\frac{1}{M} \sum_{i=1}^M c_i^2 \right) = \frac{\bar{c}^2}{n} = \frac{\bar{c}^2}{\bar{n}M} \quad (\text{D11})$$

where $n = M\bar{n}$ is the total number of samples and $\bar{c}^2$ is the average of the constants c_i^2 . In conclusion σ^2 is M times smaller than the typical variance $\bar{c}^2/\bar{n}$ of the single terms. In other words, given the desired level of accuracy in computing $\phi(R)$, i.e. the desired value of σ^2 , requiring a total number of samples n , it is not necessary to evaluate each of the M terms $\phi_i(R)$ with the same accuracy, but it is sufficient to devote to each one a number of samples equal to $\bar{n} = n/M$. This

means that the computational time necessary to evaluate $\phi(R)$ does not depend on the number m of particles. The same argument is valid also for the average in Eq. (D7), computed by means of Eq. (D8).

APPENDIX E: ERROR EVALUATION

According to the Eq. B13, the computation of the rate $\phi(R)$ of close approaches within a given distance R , using a sampling model, requires the generation of a finite number n of sample values (r_k, v_k) of the relative position and velocity, and the computation of the sum $w(R) = \sum_{k:r_k < R} v_k$. The sum contains a number $n(R)$ of terms corresponding to the number of samples for which the corresponding relative distance r_k is less than R . Both each v_k and the number of generated samples $n(R)$ are random variables. Consequently $w(R)$ and $\phi(R)$ computed by means of Eq. B13 are random variables too. Therefore, depending on the exact sequence of the random sampling, at each repeated computation, the value of $w(R)$ (and $\phi(R)$) fluctuates according to a particular parent distribution. Here we want to provide an estimation of the entity of such random fluctuations. To do this we have to study the statistical properties of the random variate $w(R)$.

Let $\psi(r, v)$ be the parent distribution the pair of random variables (r, v) , corresponding to a dynamical status of the system and generated using a suitable sampling model. Let $\psi(v|R)$ be the marginal distribution of the relative velocity for all the pairs with $r < R$:

$$\psi(v|R) = \int_0^R \psi(v, r) dr \quad (\text{E1})$$

Let us name $v(R)$ the generic value of the relative velocity for dynamical states with $r < R$. The expected value of the sum $w(R)$ is the product between the expected number of terms of the sum, i.e. the expected value of $n(R)$, and the expected value of $v(R)$:

$$\text{E}\{w(R)\} = \text{E}\{n(R)\}\text{E}\{v(R)\} \quad (\text{E2})$$

where:

$$\text{E}\{v(R)\} = \int_0^\infty v \psi(v|R) dv \quad (\text{E3})$$

Computing the square of $w(R)$:

$$w^2(R) = \sum_i v_i^2 + \sum_{i \neq j} v_i v_j \quad (\text{E4})$$

we can write:

$$\begin{aligned} \text{E}\{w^2(R)\} &= \\ &= \text{E}\{n(R)\}\text{E}\{v^2(R)\} + \text{E}\{n^2(R) - n(R)\}\text{E}\{v_i v_j\}_{i \neq j} \end{aligned} \quad (\text{E5})$$

being $n^2(R) - n(R)$ the number of terms of the second sum, and where:

$$\text{E}\{v^2(R)\} = \int_0^\infty v^2 \psi(v|R) dv \quad (\text{E6})$$

On the other hand each random sample is generated independently and no correlation exists between v_i and v_j if $i \neq j$, therefore:

$$\text{E}\{v_i v_j\}_{i \neq j} = \text{E}\{v_i\}\text{E}\{v_j\} = \text{E}^2\{v(R)\} \quad i \neq j \quad (\text{E7})$$

Substituting (E7) into (E5) we obtain:

$$\text{E}\{w^2(R)\} = \text{E}\{n(R)\}\text{Var}\{v(R)\} + \text{E}\{n^2(R)\}\text{E}^2\{v(R)\} \quad (\text{E8})$$

where $\text{Var}\{v(R)\}$ is the variance of $v(R)$:

$$\text{Var}\{v(R)\} = \text{E}\{v^2(R)\} - \text{E}^2\{v(R)\} \quad (\text{E9})$$

Finally, from (E9) and (E2), we can write the variance of $w(R)$:

$$\begin{aligned} \text{Var}\{w(R)\} &= \text{E}\{w^2(R)\} - \text{E}^2\{w(R)\} = \\ &= \text{E}\{n(R)\}\text{Var}\{v(R)\} + \text{Var}\{n(R)\}\text{E}^2\{v(R)\} \end{aligned} \quad (\text{E10})$$

We can simplify this last equation taking into account that the number $n(R)$ of samples (r, v) with $r < R$ is a binomial random variate with probability of success much lower than 1, or in other words, it is a Poisson random variable. In this case we can put $\text{E}\{n(R)\} = \text{Var}\{n(R)\}$ and the last equation becomes:

$$\text{Var}\{w(R)\} = \text{E}\{n(R)\}\text{E}\{v^2(R)\} \quad (\text{E11})$$

Once we obtained an evaluation of $\text{Var}\{w(R)\}$, we can use it to compute the variance of $\phi(R)$, or of any other parameter the definition of which contains only $w(R)$ as random variate (for example the ratio $\phi(R)/R^2$ for the computation of the intrinsic probability of collision). In particular:

$$\text{Var}\{\phi(R)\} = \frac{9}{16} \frac{p(R)}{R^2} \text{E}\{v^2(R)\} \frac{1}{n} \quad (\text{E12})$$

where $p(R)$ is the probability that r_k is smaller than R and then $\text{E}\{n(R)\} = np(R)$.

In practical terms, in a random generation of n samples (r_k, v_k) , of which $n(R)$ samples have $r_k < R$, our best estimations of $\text{E}\{n(R)\}$ and $\text{E}\{v^2(R)\}$ are:

$$\text{E}\{n(R)\} \sim n(R)$$

$$\text{E}\{v^2(R)\} \sim \frac{1}{n(R)} \sum_{k:r_k < R} v_k^2$$

then our best estimation of $\text{Var}\{w(R)\}$ is simply:

$$\text{Var}\{w(R)\} \sim \sum_{k:r_k < R} v_k^2 \quad (\text{E13})$$

The evaluation of the variance of the sample mean of a parameter q as computed from Eq. C12 with a finite number n of samples is more difficult, requiring the study of the ratio of two random variates $\sum_{k:r_k < R} q_k v_k$ and $\sum_{k:r_k < R} v_k$.

We limit ourselves only to a rough estimation. If $n(R)$ is the number of samples with $r_k < R$ and σ is the sample standard deviation of the parameter q , evaluated using the $n(R)$ values q_k , we simply put $\sigma/\sqrt{n(R)}$ as the uncertainty of the computation of the expected value of q .

The previous evaluation of the uncertainties has been numerically verified in some test cases, consisting in repeating the computations of $\phi(R)$ from Eq. (B13) and $\text{Var}\{w(R)\}$ from Eq. (E13) with independent sampling series. The random fluctuations of those values resulted fully compatible with the expected analytical estimation.

APPENDIX F: VALIDATION TESTS

In this appendix we report the results of some tests done to validate the Monte Carlo algorithm described in this paper, and in particular to check the correctness of Eqs. (B13), (C11) and (C13) using independent methods.

As test cases we consider a set of systems with canonical behavior, i.e. groups of intersecting orbits with fixed semi-major axes a , eccentricities e and inclinations I , and with longitudes of the nodes Ω and arguments of the pericenters ω circulating uniformly with time. The motion along each orbit is obviously fully described by the Kepler's second law, that is in terms of uniform variation of the mean anomaly M with time.

For each case we computed $\phi(R)/R^2$ versus R using Eq. (B13), the distribution $\Psi(R, \Delta v)$ of the relative velocity using Eq. (C11) for different values of R , and the value of $\langle v(R) \rangle$ versus R using Eq. (C13). The sampling algorithm used to generate the series of samples (r_k, v_k) consists of the steps (C1)-(C3) described in Section 3.

Moreover, for each test case we computed the mean intrinsic probability of collision P_i , the frequency distribution of the impact velocity $\Psi(U)$ and its mean value U_m , using one or more classical methods based on the Öpik/Wetherill's dynamical hypotheses. As first check, we used the method of Dell'Oro & Paolicchi (1998), in the version corresponding to the particular case of canonical statistics (see the paper Dell'Oro & Paolicchi (1998) for details). The results are then cross-checked using the independent method of Bottke et al. (1994). In this appendix we refer to the two methods of Bottke et al. (1994) and Dell'Oro & Paolicchi (1998) as “integral methods”, because they consist in computing an integral giving the value of the intrinsic probability of collision or the distribution of the impact velocity, differing from our method based on a Monte Carlo simulation. It is worthwhile to recall that our method does not provide directly P_i and U_m , but its basic outputs are the values of $\phi(R)$ and $\langle v(R) \rangle$ as functions of R , from which P_i and U_m are then estimated.

The validations consist in verifying if:

- a threshold R_ϕ can be identified such that $\phi(R)$ tends to be proportional to R^2 for $R < R_\phi$, and if the asymptotic value of the ratio $\phi(R)/R^2$ for $R \rightarrow 0$ is compatible with the value of P_i (apart from random fluctuations);
- a threshold R_v can be identified such that $\langle v(R) \rangle$ tends to be constant for $R < R_v$, and if the asymptotic value of $\langle v(R) \rangle$ for $R \rightarrow 0$ is compatible with the value of U_m (apart from random fluctuations);
- the limit distribution $\Psi(R, \Delta v)$ of the relative velocity

Table F1. List of the first ten numbered Main Belt asteroids, and the values of their orbital elements used in this paper for the validation tests.

name	a (AU)	e	I (deg)
(1) Ceres	2.76797	0.075783	10.592
(2) Pallas	2.772	0.231158	34.84
(3) Juno	2.67107	0.25583	12.987
(4) Vesta	2.36191	0.088834	7.14
(5) Astraea	2.57348	0.191188	5.369
(6) Hebe	2.42604	0.201507	14.748
(7) Iris	2.3858	0.231116	5.522
(8) Flora	2.20158	0.156688	5.887
(9) Metis	2.38657	0.122178	5.574
(10) Hygiea	3.14212	0.114692	3.838

Table F2. Collisions among the first ten numbered Main Belt asteroids (see text for details): mean intrinsic probability of collision P_i expressed in $10^{-18} \text{ km}^{-2} \text{ yr}^{-1}$, and mean impact velocity U_m in km/s, computed using the methods of Dell'Oro & Paolicchi (1998) and Bottke et al. (1994). See text for the meaning of the two cases A and B.

case	P_i	U_m
A	3.169 ± 0.002	5.217 ± 0.001
B	5.035 ± 0.002	5.910 ± 0.002

computed for small values of R is compatible with the distribution $\Psi(U)$ of the impact velocity (apart from random fluctuations). In practical terms, this test is done comparing $\Psi(U)$ with $\Psi(R, \Delta v)$ computed for a value of $R < R_v$.

F1 Multiple pairs of orbits

In Appendix D we have described how to use our Monte Carlo method in the case of systems composed of more than two particles, in order to compute the average value $\bar{\phi}(R)$ of the close encounter rate over different pairs of particles. We have shown how to avoid the two-steps procedure of calculating at first the values of $\phi(R)$ for each pair of particles and the computing the mean value of them. In fact, with our technique, it is possible to directly compute the mean value with only one sampling of the phase space of the ensemble of particles as a whole. A test concerning the close encounters among a group of different orbits has been done in order to verify the correctness of the Eqs. (D5) and (D8). The orbits used in this test are those of the first ten numbered Main Belt asteroids. The values of their orbital elements used in our computations are listed in Table F1.

We have considered two different cases:

A: rate, and corresponding distribution of the relative velocity, of close approaches between (1) Ceres and any of the other asteroids of the group;

B: rate, and corresponding distribution of the relative velocity, of close approaches between any member of the group and any other different member of the group;

In the case A nine different pairs of orbits exists, while in case B forty five combinations are possible.

Table F2 reports the values of the mean intrinsic proba-

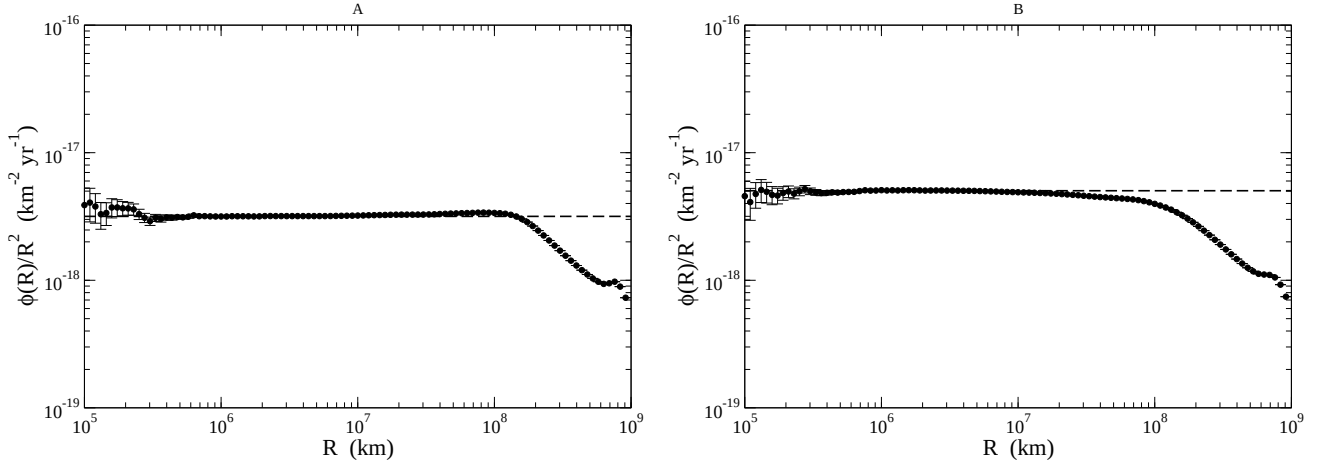


Figure F1. Close approaches among the first ten numbered Main Belt asteroids. Value of $\phi(R)/R^2$ versus R . The horizontal long-dashed line represents the value of P_i . Left: collision between (1) Ceres and the other nine asteroids (case A in the text). Right: collisions between all possible pair of orbits (case B).

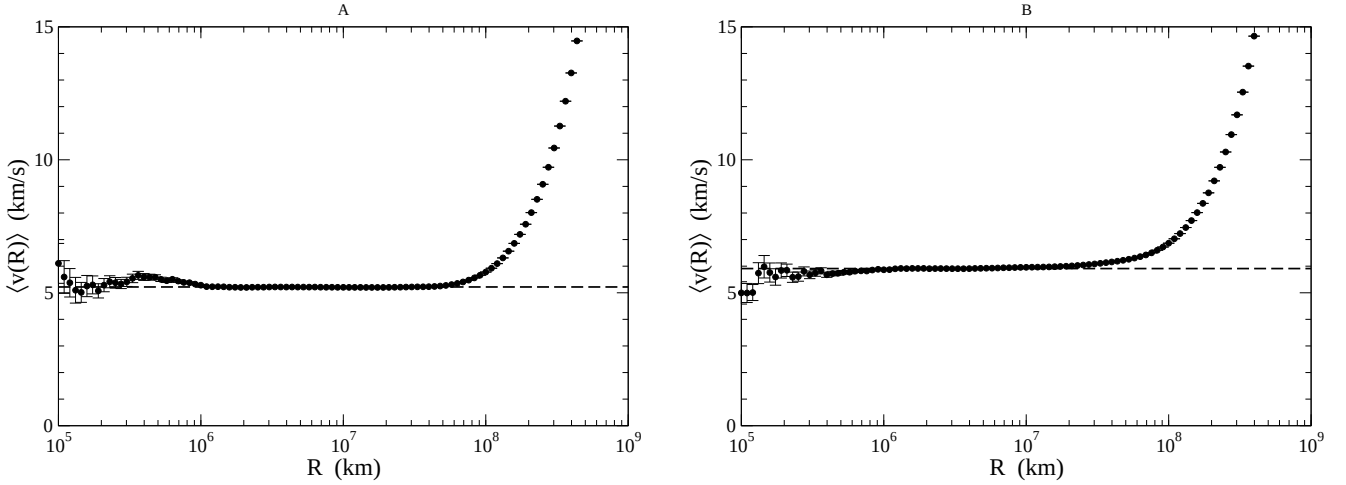


Figure F2. The mean relative velocity during close approach versus R , for the same cases of Fig. F1. The horizontal long-dashed line represents the value of U_m .

bility of collision P_i and mean impact velocity U_m , computed using the integral methods. For the two cases A and B we have computed $\phi(R)/R^2$ and $\langle v(R) \rangle$ for different values of R , using our method. The results are plotted in Fig. F1 and F2 respectively. Long-dashed horizontal lines in both figures represent the values of P_i and U_m in Table F2. In both cases the asymptotic values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ for $R < 10^7$ km are compatible with the values of P_i and U_m respectively, apart from the numerical fluctuation for values close to 10^5 km. The evidence that the mean relative velocity $\langle v(R) \rangle$ becomes constant for values of R of the order of 10^7 km suggests that the distribution of the relative velocity does not change anymore below $R \sim 10^7$ km. In Fig. F3 we have plotted the distribution of the impact velocity provided by the integral methods, drawn as a solid

line, superimposed to the distribution of the encounter velocity for close approaches within a distance less than 5×10^7 km obtained with the present method, and represented by black dots. The distributions of the impact velocity shows the characteristic peak-like features due to the contribution of the different pairs of orbits of the group, as in Fig. 3. It is difficult to reproduce with our method the details of each peak of the impact velocity distribution, requiring an excessive oversampling of the phase space and consequently a very long computational time. On the other hand the distributions obtained with the present method well reproduce the fundamental features of the impact velocity distribution.

F2 Collisions in Main Belt environment

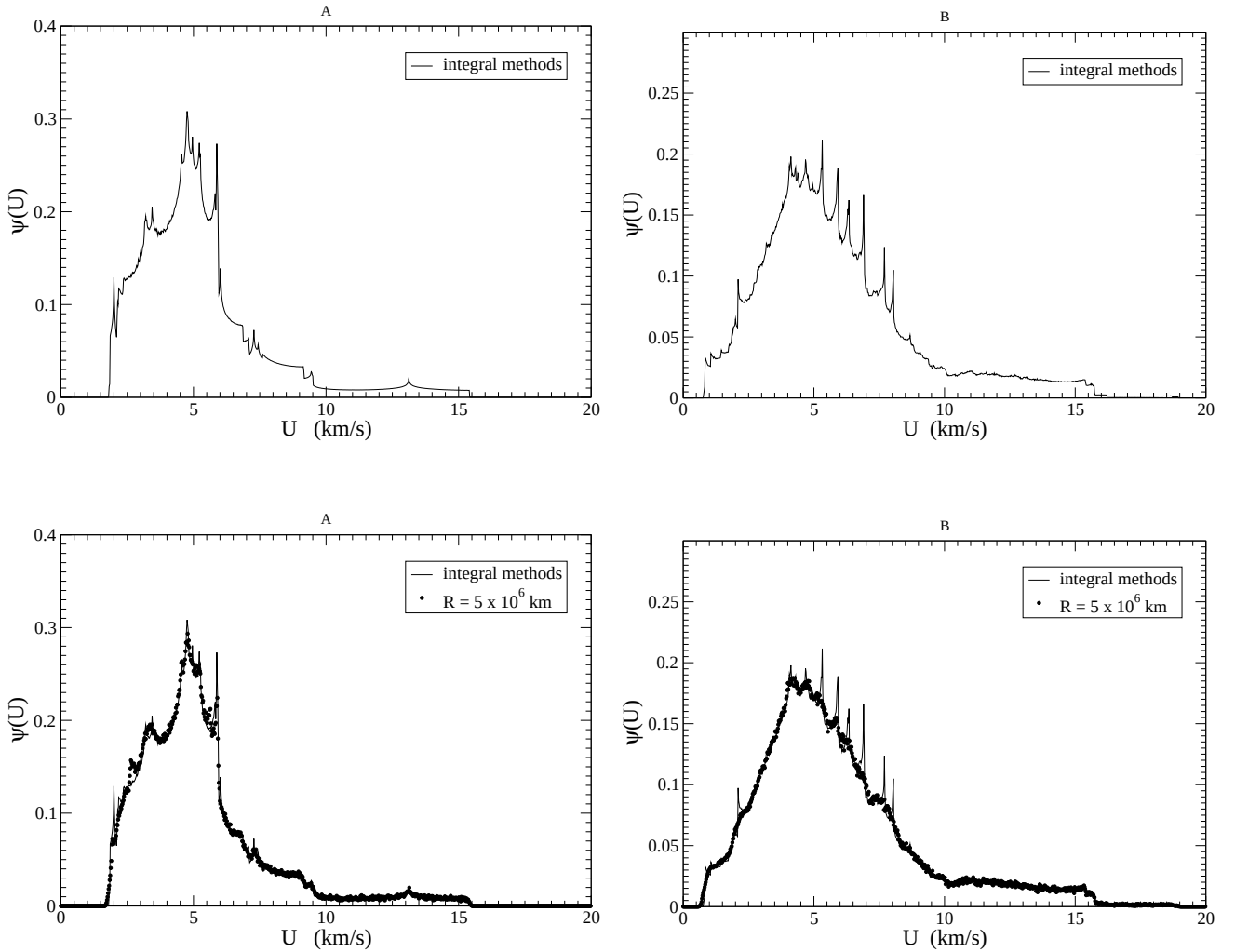


Figure F3. Distributions of the impact velocities among the first ten number Main Belt Asteroids. Solid lines: distributions of the impact velocity computed using the integral methods. Black dots: encounter velocity distribution produced by the present method for the close encounter within a distance $R = 5 \times 10^6$ km. Left: collision between (1) Ceres and the other nine asteroids. Right: collisions between all possible pair of orbits.

Finally we performed a series of tests consisting in studying the statistics of collisions between a list of targets and the whole population of the Main Belt asteroids. The targets are selected among objects having some particular scientific interest, as to be the largest remnant of an asteroid family (and therefore with an orbit probably close to that of the family's parent body) or being visited by probes during space missions. For what concerns the list of impactors, to mitigate the effect of the observational bias on the distribution of orbital elements, we selected only the Main Belt asteroids with absolute magnitude H less than 15. As shown in Fig. F4 the list of known Main Belt asteroids with $H < 15$ is probably complete, regardless the interval of semimajor axes. The list of selected projectiles includes 71066 objects. The data has been downloaded from the Minor Planet website¹ on March 17, 2016.

For the targets, we used the proper elements (Milani & Knežević 1990) from the AstDyS-2 website². The reason is that osculating elements of Main Belt asteroids have short- and long-term variations due to planetary perturbations. The probability of collision depends on the orbital elements, but we cannot know which osculating orbit the target had at the moment of collision. Proper elements, obtained filtering out secular variations, represent in some sense a sort of average orbit of the target. For projectiles we used the osculating elements instead. The reason is that the distribution of osculating elements represents a sort of snapshot of the typical dynamical configuration of the collisional environment, including the extension of its variations.

As always done in this paper, the values of P_i and U_m for each target include the contributions of all selected projectiles, regardless if the orbit of a particular projectile can

¹ URL: <http://www.minorplanetcenter.net/>

² URL: <http://hamilton.dm.unipi.it/astdys/>

Table F3. Statistics of collisions between selected targets and all Main Belt asteroids with absolute magnitude less than 15. The table lists the values of the mean intrinsic probability of collision P_i and the mean impact velocity U_m computed by means of the integral methods, compared respectively to the estimated asymptotic values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ for $R \rightarrow 0$, obtained using the method described in this work. For the parameter Δ/σ see text.

target	P_i ($10^{-18} \text{ km}^{-2} \text{ yr}^{-1}$)	$\lim_{R \rightarrow 0} \phi(R)/R^2$ ($10^{-18} \text{ km}^{-2} \text{ yr}^{-1}$)	Δ/σ	U_m (km/s)	$\lim_{R \rightarrow 0} \langle v(R) \rangle$ (km/s)	Δ/σ
(1) Ceres	3.661 ± 0.002	3.64 ± 0.02	-0.88	4.7773 ± 0.0009	4.80 ± 0.01	2.4
(2) Pallas	1.974 ± 0.003	1.95 ± 0.02	-0.88	11.719 ± 0.002	11.73 ± 0.03	0.23
(4) Vesta	3.809 ± 0.002	3.79 ± 0.02	-0.85	4.549 ± 0.001	4.551 ± 0.009	0.17
(5) Astraea	4.711 ± 0.003	4.71 ± 0.02	-0.23	4.8701 ± 0.0008	4.860 ± 0.009	-1.2
(10) Hygiea	2.698 ± 0.002	2.72 ± 0.02	1.3	4.217 ± 0.001	4.23 ± 0.01	1.3
(15) Eunomia	3.518 ± 0.003	3.54 ± 0.02	0.97	5.670 ± 0.001	5.66 ± 0.01	-0.45
(20) Massalia	5.117 ± 0.003	5.10 ± 0.02	-0.59	4.4114 ± 0.0009	4.429 ± 0.008	2.2
(21) Lutetia	4.846 ± 0.003	4.83 ± 0.02	-0.81	4.208 ± 0.001	4.224 ± 0.008	2.1
(24) Themis	3.364 ± 0.002	3.36 ± 0.02	-0.35	4.020 ± 0.001	4.056 ± 0.008	4.4
(93) Minerva	3.708 ± 0.003	3.69 ± 0.02	-0.68	4.839 ± 0.001	4.85 ± 0.01	1.5
(135) Hertha	5.112 ± 0.004	5.10 ± 0.02	-0.45	4.544 ± 0.001	4.571 ± 0.008	3.3
(145) Adeona	3.575 ± 0.003	3.58 ± 0.02	0.30	5.498 ± 0.001	5.51 ± 0.01	1.1
(158) Koronis	4.382 ± 0.003	4.35 ± 0.02	-1.8	3.465 ± 0.001	3.482 ± 0.006	2.7
(170) Maria	3.353 ± 0.003	3.36 ± 0.02	0.19	6.061 ± 0.001	6.07 ± 0.01	0.70
(221) Eos	3.038 ± 0.002	3.03 ± 0.02	-0.31	4.4392 ± 0.0008	4.45 ± 0.01	1.4
(243) Ida	4.423 ± 0.002	4.38 ± 0.02	-2.1	3.4624 ± 0.0008	3.477 ± 0.006	2.3
(253) Mathilde	4.157 ± 0.003	4.17 ± 0.02	0.73	5.2546 ± 0.0008	5.26 ± 0.01	0.82
(434) Hungaria	1.346 ± 0.004	1.37 ± 0.02	1.7	8.961 ± 0.007	9.00 ± 0.03	1.2
(490) Veritas	2.30 ± 0.04	2.27 ± 0.02	-0.80	4.255 ± 0.008	4.26 ± 0.01	0.031
(668) Dora	3.601 ± 0.002	3.60 ± 0.02	-0.21	5.1605 ± 0.0007	5.16 ± 0.01	-0.38
(847) Agnia	4.296 ± 0.002	4.29 ± 0.02	-0.26	3.7702 ± 0.0008	3.773 ± 0.007	0.42
(951) Gaspra	3.720 ± 0.004	3.70 ± 0.02	-0.73	4.848 ± 0.002	4.85 ± 0.01	0.048
(1040) Klumpkea	2.040 ± 0.002	2.03 ± 0.02	-0.46	6.791 ± 0.001	6.82 ± 0.02	1.5
(1726) Hoffmeister	4.188 ± 0.002	4.19 ± 0.02	0.24	3.7540 ± 0.0009	3.768 ± 0.007	1.9
(2076) Levin	3.860 ± 0.003	3.86 ± 0.02	0.055	4.722 ± 0.001	4.74 ± 0.01	2.2
(2867) Steins	3.386 ± 0.002	3.38 ± 0.02	-0.38	5.118 ± 0.001	5.10 ± 0.01	-1.4

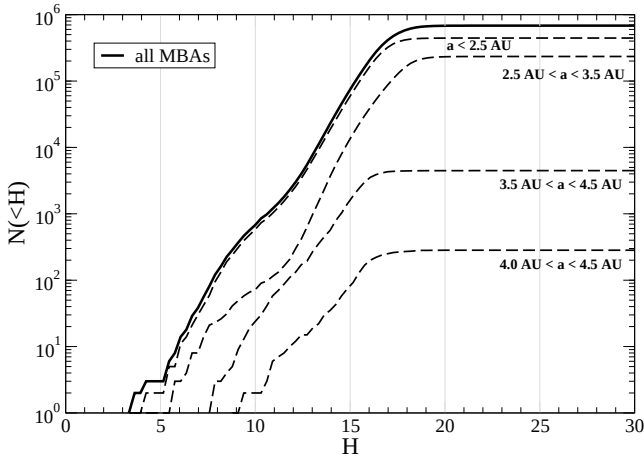


Figure F4. Cumulative distribution of the absolute magnitudes of the Main Belt asteroids (data from Minor Planet Center, March 2016). Bold line represents the distribution of all MBAs, while long-dashed lines the distribution for some sub-groups, selected according to different intervals of semimajor axis.

cross the orbit of the target or not. In the second case the projectile contributes with a null term to the average value P_i . The advantage of this approach is that the mean num-

ber N of impacts undergone by a target of diameter D_0 with Main Belt asteroids during a time span T can be written as:

$$N = \frac{1}{4} P_i T \int (D_0 + D)^2 F(D) dD \quad (\text{F1})$$

where $F(D)$ is the size distribution of the overall population, without worrying about which asteroids actually can collide on target or not.

The results of this test are shown in Table F3. For each selected target the mean intrinsic probability of collision P_i and the mean impact speed U_m obtained using the integral methods are reported, together with the corresponding estimated limit values of $\phi(R)/R^2$ and $\langle v(R) \rangle$ for $R \rightarrow 0$, obtained using our method. The parameter Δ/σ is defined as follows: if x' is the value provided by the present method and σ' its error, and x_0 and σ_0 are respectively the value of the same quantity and its error computed using the integral methods, then:

$$\frac{\Delta}{\sigma} = \frac{x' - x_0}{\sqrt{\sigma'^2 + \sigma_0^2}} \quad (\text{F2})$$

The value of the ratio Δ/σ tells us how much the two values x' and x_0 are statistically different. If the two methods are equivalent we expect that Δ/σ is of the order of few units. The Table F3 confirms the mutual consistency of the two methods. In some cases for U_m and the corresponding $\lim_{R \rightarrow 0} \langle v(R) \rangle$, like (24) Themis, the value of Δ/σ results to

be rather large. This is due to the fact that our rough determination of the error of $\langle v(R) \rangle$ (see Appendix E) can in some cases underestimate the real size of numerical fluctuations. This has been confirmed repeating the computation of the values in Table F3. Moreover, the values of $\lim_{R \rightarrow 0} \langle v(R) \rangle$ computed by the our method seem systematically slightly larger than the corresponding U_m . This depends on the values of R we use to estimate $\lim_{R \rightarrow 0} \langle v(R) \rangle$ and on the fact that the function $\langle v(R) \rangle$ versus R is perfectly constant or not in this range of values of R . In general $\langle v(R) \rangle$ decreases decreasing R , and the limit value of $\langle v(R) \rangle$ for $R \rightarrow 0$ is thus always smaller than all other values. A similar situation, although less evident in Table F3, occurs for P_i : in general, but not always, $\phi(R)/R^2$ increases decreasing R . Therefore its limit value for $R \rightarrow 0$ is, in this case, an upper limit, resulting slightly underestimated.

This paper has been typeset from a $\text{\TeX}/\text{\LaTeX}$ file prepared by the author.